

Free Vibration Analysis of Centrifugally Stiffened Non Uniform Timoshenko Beams

Diana V. Bambill, Daniel H. Felix, Raúl E. Rossi and Alejandro R. Ratazzi
*Universidad Nacional del Sur, UNS, Departamento de Ingeniería,
Instituto de Mecánica Aplicada, IMA,
Consejo Nacional de Investigaciones Científicas y Técnicas, CONICET,
Argentina*

1. Introduction

Rotating beams – like structures are widely used in many engineering fields and are of great interest as they can be used to model blades of wind turbines, helicopter rotors, robotic manipulators, turbo-machinery and aircraft propellers. The governing differential equations of motion in free vibration of a non-uniform rotating Timoshenko beam, with general elastic restraints at the ends are solved using the differential quadrature method, (Bellman & Roth, 1986; Felix et al., 2008, 2009). The equations of motion are derived to include the effects of shear deformation, rotary inertia, hub radius, ends elastically restrained and non-uniform variation of the cross-sectional area of the beam. The presence of a centrifugal force due to the rotational motion is considered as Banerjee has developed, using Hamilton's principle to capture the centrifugal stiffening arising in fast rotating structures, (Banerjee, 2001). With the proposed model, a great number of different situations are admitted to be solved. Particular cases with classical restraints can be deduced for limiting values of the rigidities. Also step changes in cross-section are considered (Naguleswaran, 2004).

The natural vibration frequencies and mode shapes of rotating beams have been a topic of interest and have received considerable attention. A large number of researchers have studied the dynamic behavior of rotating uniform or tapered Euler-Bernoulli beams. (Yang et al., 2004; Özdemir & Kaya, 2006; Lin & Hsiao, 2001). Banerjee derived the dynamic stiffness matrix of a rotating Bernoulli-Euler beam using the Frobenius method of solution in power series and he includes the presence of an axial force at the outboard end of the beam in addition to the existence of the usual centrifugal (Banerjee, 2000).

Not so many studies have tackled the problem of rotating beams taking into account rotary inertia, shear deformation and their combined effects, hub radius and ends elastically restrained, (Bambill et al., 2010). In applications where the rotary inertia and the shear deformation effects are not significant, an analysis based on the Euler-Bernoulli beam theory can be used. However, Timoshenko theory allows describing the vibration of short beams, sandwich composite beams or high modes of a slender beam, (Rossi et al., 1991; Seon et al., 1999). (Banerjee et al., 2006) investigated the free bending vibration of rotating tapered Timoshenko beams by the dynamic stiffness method. (Ozgumus & Kaya, 2010) used the Differential Transform Method for free vibration analysis of a rotating, tapered Timoshenko beam.

The finite element method was used by (Hodges & Rutkowski, 1981). (Vinod et al., 2007) presented a study about spectral finite element formulation for a rotating beam subjected to small duration impact. (Gunda & Ganguli, 2008) developed a new beam finite element whose basis functions were obtained by the exact solution of the governing static homogenous differential equation of a stiff string, which resulted from an approximation in the rotating beam equation. (Singh et al., 2007) used the Genetic Programming to create an approximate model of rotating beams. (Gunda et al., 2007) introduced a low degree of freedom model for dynamic analysis of rotating tapered beams based on a numerically efficient superelement, developed using a combination of polynomials and Fourier series as shape functions. (Kumar & Ganguli, 2009) looked for rotating beams whose eigenpair, frequency and mode-shape, is the same as that of uniform non rotating beams for a particular mode. An interesting paper (Ganesh & Ganguli, 2011) presented physics based basis function for vibration analysis of high speed rotating beams using the finite element method. The basis function gave rise to shape functions which depend on position of the element in the beam, material, geometric properties and rotational speed of the beam.

The present study tries to provide not only solutions for practical engineering situations but they also may be useful as benchmark for comparing other numerical models. The proposed differential quadrature method, offers a useful and accurate procedure for the solution of linear and non linear partial differential equations. It was used by Bellman in the 1970's. He used this method to calculate the natural frequencies of transverse vibration of a rotating cantilever beam. (Bellman & Casti, 1971). Other authors have used the differential quadrature method and recognized it as an effective technique for solving this kind of problems, (Bert & Malik, 1996; Shu & Chen, 1999; Choi et al., 2000; Liu & Wu, 2001; Shu, 2000).

Numerical results are obtained for the natural frequencies of transverse vibration and the mode shapes of rotating beams considering the elastic restraints, with non uniform variation of the cross-sectional area. Some of those cases have also been solved using the finite element method, and the sets of results are in excellent agreement.

2. Theory

Figure 1 shows the rotating tapered beam considered in the present paper. The beam could have step jumps in cross section and rotates at speed $\bar{\eta}$. The \bar{X} -axis coincides with the centroidal axis of the beam, the \bar{Y} -axis is parallel with the axis of rotation and the \bar{Z} -axis lies in the plane of rotation. L is the length of the beam, L_k is the length of the segment k and L_d is the length of the last segment of the beam. The displacement in the \bar{Y} direction is denoted as \bar{w} and the section rotation is denoted as $\bar{\psi}$. Only displacements in the $\bar{X}-\bar{Y}$ plane are taken into account and the Coriolis effects are not considered.

The centrifugal force of a beam element at a distance $\bar{R}_k + \bar{x}_k$ from the axis of rotation can be expressed as

$$d\bar{F}_k = \bar{\eta}^2 (\bar{R}_k + \bar{x}_k) dm \quad (1)$$

where $dm = \rho A_k(\bar{x}_k) d\bar{x}_k$ is its mass, with ρ the mass density of material, and $A_k(\bar{x}_k)$, is the cross-sectional area at \bar{x}_k . Figure 2. The centrifugal force $\bar{N}_k(\bar{x}_k)$ generated by $\bar{\eta}$ is

$$d\bar{N}_k(\bar{x}_k) = \bar{\eta}^2 \rho (\bar{R}_k + \bar{x}_k) A_k(\bar{x}_k) d\bar{x}_k \quad (2)$$

The total axial force at the cross section located at $\bar{R}_k + \bar{x}_k$ is

$$\bar{N}_k(\bar{x}_k) = \bar{\eta}^2 \rho \int_{\bar{x}_k}^{L_k} (\bar{R}_k + \bar{x}_k) A_k(\bar{x}_k) d\bar{x}_k + \bar{F}_{k+1} = \bar{\eta}^2 \rho \left(\bar{R}_k \int_{\bar{x}_k}^{L_k} A_k(\bar{x}_k) d\bar{x}_k + \int_{\bar{x}_k}^{L_k} A_k(\bar{x}_k) \bar{x}_k d\bar{x}_k \right) + \bar{F}_{k+1} \quad (3)$$

\bar{F}_{k+1} is the outboard force at the end of the segment k , due to the adjacent segments $k+1$ to d .

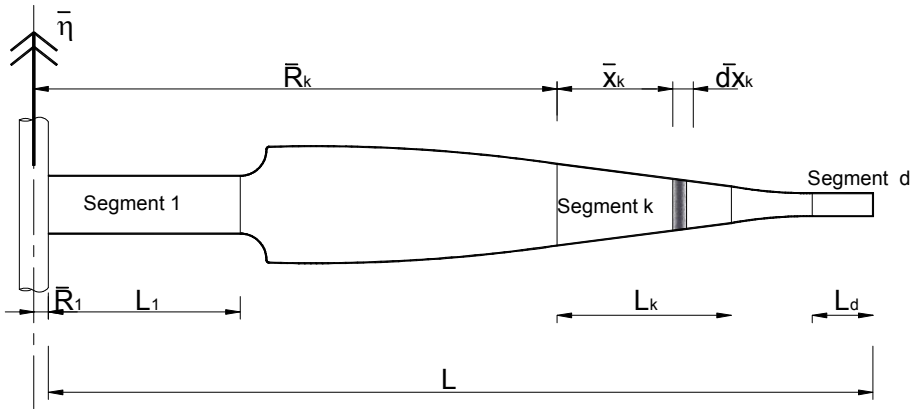


Fig. 1. Rotating beam model

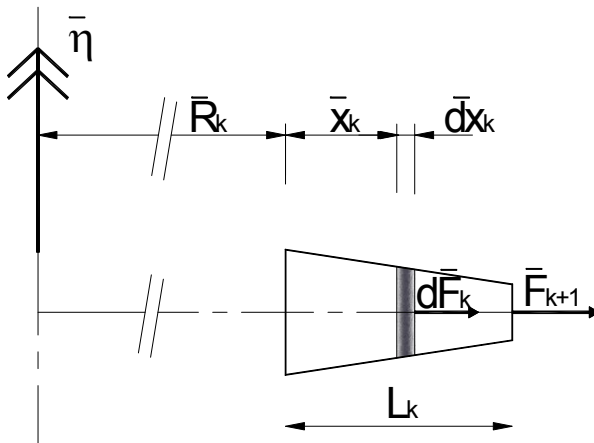


Fig. 2. Rotating beam segment k of length L_k

Finally, the tensile force can be written as

$$\bar{N}_k(\bar{x}_k) = \bar{\eta}^2 \rho \left(\bar{R}_k V_k(L_k) + \Phi_k(L_k) - \bar{R}_k V_k(\bar{x}_k) - \Phi_k(\bar{x}_k) \right) + \bar{F}_{k+1} \quad (4)$$

with

$$V_k(\bar{x}_k) = \int_0^{\bar{x}_k} A_k(\bar{x}_k) d\bar{x}_k ; \Phi_k(\bar{x}_k) = \int_0^{\bar{x}_k} A_k(\bar{x}_k) \bar{x}_k d\bar{x}_k \quad (5a,b)$$

The expressions for shear force and bending moment at an instant t in the rotating beam are

$$\bar{Q}_k^*(\bar{x}_k, t) = \bar{N}_k(\bar{x}_k) \frac{\partial \bar{w}_k(\bar{x}_k, t)}{\partial \bar{x}_k} + \kappa GA_k(\bar{x}_k) \left(\frac{\partial \bar{w}_k(\bar{x}_k, t)}{\partial \bar{x}_k} - \bar{\psi}_k(\bar{x}_k, t) \right) \quad (6)$$

$$\bar{M}_k^*(\bar{x}_k, t) = EI_k(\bar{x}_k) \frac{\partial \bar{\psi}_k(\bar{x}_k, t)}{\partial \bar{x}_k} \quad (7)$$

where $I_k(\bar{x}_k)$ is the second moment of area of the beam cross-section; t the time; $\bar{w}_k(\bar{x}_k, t)$ the transverse displacement; $\bar{\psi}_k(\bar{x}_k, t)$ the section rotation; E the Young's modulus; ν the Poisson's ratio; $G = E / 2(1 + \nu)$ the shear modulus and κ is the shear factor.

The governing differential equations of motion of a rotating Timoshenko beams (Banerjee, 2001) are:

$$\frac{\partial \bar{Q}_k^*(\bar{x}_k, t)}{\partial \bar{x}_k} = \rho A_k(\bar{x}_k) \frac{\partial^2 \bar{w}_k(\bar{x}_k, t)}{\partial t^2} \quad (8a,b)$$

$$\bar{Q}_k^*(\bar{x}_k, t) - N_k(\bar{x}_k) \frac{\partial \bar{w}_k(\bar{x}_k, t)}{\partial \bar{x}_k} + \frac{\partial \bar{M}_k^*(\bar{x}_k, t)}{\partial \bar{x}_k} + \rho I_k(\bar{x}_k) \bar{\eta}^2 \bar{\psi}_k(\bar{x}_k, t) = \rho I_k(\bar{x}_k) \frac{\partial^2 \bar{\psi}_k(\bar{x}_k, t)}{\partial t^2}$$

Assuming simple harmonic oscillation

$$\bar{w}_k(\bar{x}_k, t) = \bar{W}_k(\bar{x}_k) e^{i\omega t} ; \bar{\psi}_k(\bar{x}_k, t) = \bar{\Psi}_k(\bar{x}_k) e^{i\omega t} \quad (9a,b)$$

where ω is the circular frequency in radian per second.

The bending moment and the shear force are expressed as

$$\bar{Q}_k^*(\bar{x}_k, t) = \bar{Q}_k(\bar{x}_k) e^{i\omega t} ; \bar{M}_k^*(\bar{x}_k, t) = \bar{M}_k(\bar{x}_k) e^{i\omega t} \quad (10a,b)$$

where

$$\bar{Q}_k(\bar{x}_k) = (\bar{N}_k(\bar{x}_k) + \kappa GA_k(\bar{x}_k)) \frac{d\bar{W}_k(\bar{x}_k)}{d\bar{x}_k} - \kappa GA_k(\bar{x}_k) \bar{\Psi}_k(\bar{x}_k) ; \bar{M}_k(\bar{x}_k) = EI_k(\bar{x}_k) \frac{d\bar{\Psi}_k(\bar{x}_k)}{d\bar{x}_k} \quad (11a,b)$$

Substituting equations (9-10) into equations (8), the equations of motion for the free vibration of the segment k of the rotating beam result in:

$$-\frac{d\bar{Q}_k(\bar{x}_k)}{d\bar{x}_k} = \rho A_k(\bar{x}_k) \omega^2 \bar{W}_k(\bar{x}_k) \quad (12a,b)$$

$$-\bar{Q}_k(\bar{x}_k) + \bar{N}_k(\bar{x}_k) \frac{d\bar{W}_k(\bar{x}_k)}{d\bar{x}_k} - \frac{d\bar{M}_k(\bar{x}_k)}{d\bar{x}_k} - \rho I_k(\bar{x}_k) \bar{\eta}^2 \bar{\Psi}_k(\bar{x}_k) = \rho I_k(\bar{x}_k) \omega^2 \bar{\Psi}_k(\bar{x}_k)$$

Replacing equations (11) into equations (12), the differential equations of motion become:

$$\begin{aligned} &-\frac{d\bar{N}_k(\bar{x}_k)}{d\bar{x}_k} \frac{d\bar{W}_k(\bar{x}_k)}{d\bar{x}_k} - \bar{N}_k(\bar{x}_k) \frac{d^2\bar{W}_k(\bar{x}_k)}{d\bar{x}_k^2} - \kappa G A_k(\bar{x}_k) \left(\frac{d^2\bar{W}_k(\bar{x}_k)}{d\bar{x}_k^2} - \frac{d\bar{\Psi}_k(\bar{x}_k)}{d\bar{x}} \right) - \\ &\kappa G \frac{dA_k(\bar{x}_k)}{d\bar{x}_k} \left(\frac{d\bar{W}_k(\bar{x}_k)}{d\bar{x}_k} - \bar{\Psi}_k(\bar{x}_k) \right) = \rho A_k(\bar{x}_k) \omega^2 \bar{W}_k(\bar{x}_k) \\ &- \kappa G A_k(\bar{x}_k) \left(\frac{d\bar{W}_k(\bar{x}_k)}{d\bar{x}_k} - \bar{\Psi}_k(\bar{x}_k) \right) - EI_k(\bar{x}_k) \frac{d^2\bar{\Psi}_k(\bar{x}_k)}{d\bar{x}_k^2} - \\ &E \frac{dI_k(\bar{x}_k)}{d\bar{x}_k} \frac{d\bar{\Psi}_k(\bar{x}_k)}{d\bar{x}_k} - \rho I_k(\bar{x}_k) \bar{\eta}^2 \bar{\Psi}_k(\bar{x}_k) = \rho I_k(\bar{x}_k) \omega^2 \bar{\Psi}_k(\bar{x}_k) \end{aligned} \tag{13a,b}$$

The term $\rho I_k(\bar{x}_k) \bar{\eta}^2 \bar{\Psi}_k(\bar{x}_k)$ included in equation (13.b) was introduced by Banerjee, 2001. This term generates more realistic results especially for high rotational speeds, $\bar{\eta}^2$.

The conditions for displacements and forces between adjacent segments, k and $k+1$, are:

$$\bar{W}_k(L_k) - \bar{W}_{k+1}(0) = 0; \quad \bar{\Psi}_k(L_k) - \bar{\Psi}_{k+1}(0) = 0 \tag{14a,b}$$

$$\bar{Q}_k(L_k) - \bar{Q}_{k+1}(0) = 0; \quad \bar{M}_k(L_k) - \bar{M}_{k+1}(0) = 0 \tag{15a,b}$$

Figure 3 shows the beam elastically restrained at both ends.

The boundary conditions of the beam at its ends are, for the first segment $k=1$, at $\bar{x}_1 = 0$:

$$\bar{Q}_1(0) - \bar{K}_{W1} \bar{W}_1(0) = 0; \quad \bar{M}_1(0) - \bar{K}_{\Psi1} \bar{\Psi}_1(0) = 0 \tag{16a,b}$$

and for the last segment $k=d$, at $\bar{x}_d = L_d$:

$$\bar{Q}_d(L_d) - \bar{K}_{Wd} \bar{W}_d(0) = 0; \quad \bar{M}_d(L_d) - \bar{K}_{\Psi d} \bar{\Psi}_d(0) = 0 \tag{17a,b}$$

The four spring constants are denoted as: $\bar{K}_{W1}, \bar{K}_{Wd}, \bar{K}_{\Psi1}, \bar{K}_{\Psi d}$.

The expressions and parameters in dimensionless form are defined as follows:

$$\begin{aligned} \Omega^2 &= \frac{\rho A_1(0)}{EI_1(0)} L^4 \omega^2; \quad \eta^2 = \frac{\rho A_1(0)}{EI_1(0)} L^4 \bar{\eta}^2; \\ x &= \frac{\bar{x}_k}{L_k}; \quad l_k = \frac{L_k}{L}; \quad r_k^2 = \frac{I_k(0)}{A_k(0)}; \quad s_k = \frac{L}{r_k}; \quad R_k = \frac{\bar{R}_k}{L}; \quad W_k(x) = \frac{\bar{W}_k(\bar{x}_k)}{L_k}; \quad \Psi_k(x) = \bar{\Psi}_k(\bar{x}_k); \\ a_k(x) &= \frac{A_k(\bar{x}_k)}{A_k(0)}; \quad b_k(x) = \frac{I_k(\bar{x}_k)}{I_k(0)}; \quad v_k(x) = \frac{V_k(\bar{x}_k)}{l_k A_k(0)}; \quad \phi_k(x) = \frac{\Phi_k(\bar{x}_k)}{l_k^2 A_k(0)}; \\ N_{k+1} &= \frac{\bar{F}_{k+1}}{EA_k(0)}; \quad N_k(x) = \frac{\bar{N}_k(\bar{x}_k)}{EA_k(0)}; \quad Q_k(x) = \frac{\bar{Q}_k(\bar{x}_k)}{EA_k(0)}; \quad M_k(x) = \frac{I_k}{EI_k(0)} \bar{M}_k(\bar{x}_k); \end{aligned}$$

$$\text{and } K_{Wj} = \bar{K}_{Wj} \frac{L}{EA_1(0)} ; K_{\Psi j} = \bar{K}_{\Psi j} \frac{L}{EI_1(0)} ; \text{ with } j=1 \text{ or } j=d.$$

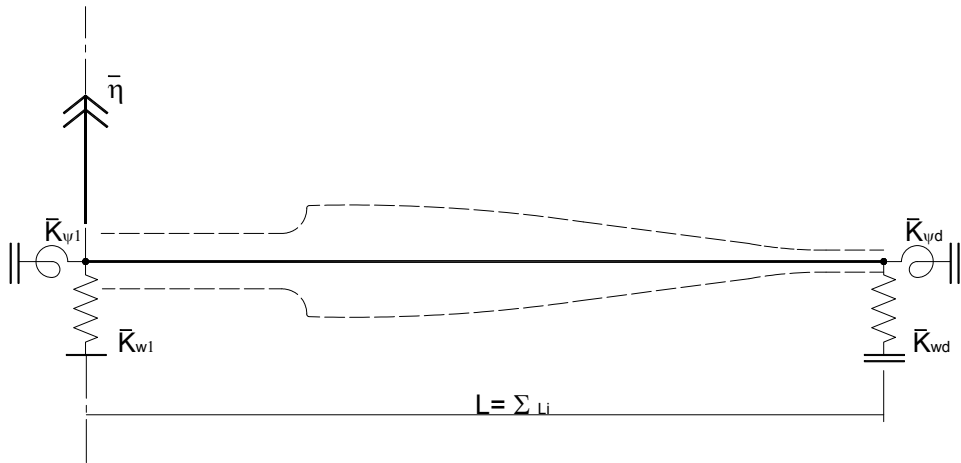


Fig. 3. Elastic restraints of the rotating beam

In each segment k of the beam, x varies between 0 and 1.

The axial force, the shear force and the bending moment in the adimensional form become:

$$N_k(x) = \eta^2 \frac{l_k^2}{s_1^2} (R_k v_k(1) + \phi_k(1) - R_k v_k(x) - \phi_k(x)) + N_{k+1} ; \text{ with } s_1 = \frac{L}{r_1} ; r_1^2 = \frac{I_1(0)}{A_1(0)} \quad (18)$$

$$Q_k(x) = \left(N_k(x) + \frac{\kappa}{2(1+\nu)} a_k(x) \right) \frac{dW_k(x)}{dx} - \frac{\kappa}{2(1+\nu)} a_k(x) \Psi_k(x) \quad (19)$$

$$M_k(x) = b_k(x) \frac{d\Psi_k(x)}{dx} \quad (20)$$

And the equations of motion in dimensionless form are:

$$\eta^2 a_k(x) (R_k + x) \frac{dW_k(x)}{dx} - \frac{s_1^2}{l_k^2} N_k(x) \frac{d^2W_k(x)}{dx^2} - \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} a_k(x) \left(\frac{d^2W_k(x)}{dx^2} - \frac{d\Psi_k(x)}{dx} \right) - \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} \frac{da_k(x)}{dx} \left(\frac{dW_k(x)}{dx} - \Psi_k(x) \right) = \Omega^2 a_k(x) W_k(x) \quad (21)$$

$$-s_1^2 \frac{\kappa}{2(1+\nu)} s_k^2 a_k(x) \left(\frac{dW_k(x)}{dx} - \Psi_k(x) \right) - \frac{s_1^2}{l_k^2} b_k(x) \frac{d^2\Psi_k(x)}{dx^2} - \frac{s_1^2}{l_k^2} \frac{db_k(x)}{dx} \frac{d\Psi_k(x)}{dx} - \eta^2 b_k(x) \Psi_k(x) = \Omega^2 b_k(x) \Psi_k(x) \quad (22)$$

The equations (14), which satisfy continuity of displacement and rotation, can be expressed in dimensionless form as follows:

$$l_k W_k(1) - l_{k+1} W_{k+1}(0) = 0; \Psi_k(1) - \Psi_{k+1}(0) = 0 \tag{23a,b}$$

and the equations (15) of compatibility of the bending moment and the shear force, result in the following adimensional equations:

$$\alpha_k Q_k(1) - \alpha_{k+1} Q_{k+1}(0) = 0; \frac{\beta_k}{l_k} M_k(1) - \frac{\beta_{k+1}}{l_{k+1}} M_{k+1}(0) = 0$$

or

$$\begin{aligned} & \alpha_k \left[\left(N_k(x) + \frac{\kappa}{2(1+\nu)} a_k(x) \right) \frac{dW_k(x)}{dx} - \frac{\kappa}{2(1+\nu)} a_k(x) \Psi_k(x) \right] \Big|_{x=1} - \\ & - \alpha_{k+1} \left[\left(N_{k+1}(x) + \frac{\kappa}{2(1+\nu)} a_{k+1}(x) \right) \frac{dW_{k+1}(x)}{dx} - \frac{\kappa}{2(1+\nu)} a_{k+1}(x) \Psi_{k+1}(x) \right] \Big|_{x=0} = 0; \tag{24a,b} \\ & \frac{\beta_k}{l_k} b_k(x) \frac{d\Psi_k(x)}{dx} \Big|_{x=1} - \frac{\beta_{k+1}}{l_{k+1}} b_{k+1}(x) \frac{d\Psi_{k+1}(x)}{dx} \Big|_{x=0} = 0 \end{aligned}$$

where $\alpha_k = \frac{A_k(0)}{A_1(0)}$; $\beta_k = \frac{I_k(0)}{I_1(0)}$.

The boundary conditions at the end closest to the axis of rotation, segment 1, $x=0$, are:

$$\begin{aligned} Q_1(0) - K_{W1} l_1 W_1(0) = 0; & \left(N_1(0) + \frac{\kappa a_1(0)}{2(1+\nu)} \right) \frac{dW_1(x)}{dx} \Big|_{x=0} - \frac{\kappa a_1(0)}{2(1+\nu)} \Psi_1(0) - K_{W1} l_1 W_1(0) = 0 \\ M_1(0) - K_{\Psi1} l_1 \Psi_1(0) = 0; & b_1(0) \frac{d\Psi_1(x)}{dx} \Big|_{x=0} - K_{\Psi1} l_1 \Psi_1(0) = 0 \end{aligned} \tag{25a,b}$$

and at the other end of the rotating beam, segment d , $x=1$, they are:

$$\begin{aligned} Q_d(1) - K_{Wd} \frac{l_d}{\alpha_d} W_d(1) = 0; & \left(N_d(1) + \frac{\kappa a_d(1)}{2(1+\nu)} \right) \frac{dW_d(x)}{dx} \Big|_{x=1} - \frac{\kappa a_d(1)}{2(1+\nu)} \Psi_d(1) - \frac{K_{Wd} l_d}{\alpha_d} W_d(1) = 0 \\ M_d(1) - \frac{K_{\Psi d} l_d}{\beta_d} \Psi_d(1) = 0; & b_d(1) \frac{d\Psi_d(x)}{dx} \Big|_{x=1} - \frac{K_{\Psi d} l_d}{\beta_d} \Psi_d(1) = 0 \end{aligned} \tag{26a,b}$$

where $N_d(1)$ is an outboard force at the end of the beam, farthest from the axis of rotation, that is equal to zero in the present study.

3. Differential Quadrature Method, DQM

In order to obtain the DQM analog equations from the governing equations of the rotating beam, the beam segment domain is discretized in a grid of i points, using the Chebyshev - Gauss - Lobato expression, (Shu, 2000). (See Fig. A.1 in Appendix A)

Equations (18, 19, 20) assumed the form:

$$N_k(x_i) = \eta^2 \frac{l_k^2}{s_1^2} (R_k v_k(1) + \phi_k(1) - R_k v_k(x_i) - \phi_k(x_i)) + N_{k+1} \quad (27)$$

$$Q_k(x_i) = \left(N_k(x_i) + \frac{\kappa}{2(1+\nu)} a_k(x_i) \right) \sum_{j=1}^n A_{ij}^{(1)} W_{kj} - \frac{\kappa}{2(1+\nu)} a_k(x_i) \Psi_{ki} \quad (28)$$

$$M_k(x_i) = b_k(x_i) \sum_{j=1}^n A_{ij}^{(1)} \Psi_{kj} \quad (29)$$

The equations of motion (21) and (22) become:

$$\begin{aligned} & \left(\eta^2 a_k(x_i) (R_k + x_i) - \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} \frac{da_k(x_i)}{dx} \right) \sum_{j=1}^n (A_{ij}^{(1)}) W_{kj} - \\ & - \left(\frac{s_1^2}{l_k^2} N_k(x_i) + \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} a_k(x_i) \right) \sum_{j=1}^n (A_{ij}^{(2)}) W_{kj} + \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} a_k(x_i) \sum_{j=1}^n A_{ij}^{(1)} \Psi_{kj} + \\ & + \frac{\kappa}{2(1+\nu)} \frac{s_1^2}{l_k^2} \frac{da_k(x_i)}{dx} \Psi_{ki} = \Omega^2 a_k(x_i) W_{ki} \end{aligned} \quad (30)$$

$$\begin{aligned} & - \frac{\kappa}{2(1+\nu)} s_1^2 s_k^2 a_k(x_i) \sum_{j=1}^n A_{ij}^{(1)} W_{kj} - \frac{s_1^2}{l_k^2} b_k(x_i) \sum_{j=1}^n A_{ij}^{(2)} \Psi_{kj} + \\ & + \left(\frac{\kappa}{2(1+\nu)} s_1^2 s_k^2 a_k(x_i) - \eta^2 b_k(x_i) \right) \Psi_{ki} - \frac{s_1^2}{l_k^2} \frac{db_k(x_i)}{dx} \sum_{j=1}^n A_{ij}^{(1)} \Psi_{kj} = \Omega^2 b_k(x_i) \Psi_{ki} \end{aligned} \quad (31)$$

where the $A_{ij}^{(1)}$ and $A_{ij}^{(2)}$ are the weighting coefficients of linear algebraic equations. (See Appendix A.1 for more details).

Finally, the conditions (23) and (24) are replaced by:

$$l_k W_{kn} - l_{k+1} W_{(k+1)1} = 0; \quad \Psi_{kn} - \Psi_{(k+1)1} = 0; \quad (32a,b)$$

$$\begin{aligned} & \alpha_k \left(\left(N_k(1) + \frac{\kappa}{2(1+\nu)} a_k(1) \right) \sum_{j=1}^n A_{nj}^{(1)} W_{kj} - \frac{\kappa}{2(1+\nu)} a_k(1) \Psi_{kn} \right) \\ & - \alpha_{k+1} \left(\left(N_{k+1}(0) + \frac{\kappa}{2(1+\nu)} a_{k+1}(0) \right) \sum_{j=1}^n A_{1j}^{(1)} W_{(k+1)j} - \frac{\kappa}{2(1+\nu)} a_{k+1}(0) \Psi_{k1} \right) = 0; \end{aligned} \quad (33a,b)$$

$$\frac{\alpha_k}{l_k} b_k(1) \sum_{j=1}^n A_{nj}^{(1)} \Psi_{kj} - \frac{\alpha_{k+1}}{l_{k+1}} b_{k+1}(0) \sum_{j=1}^n A_{1j}^{(1)} \Psi_{(k+1)j} = 0$$

and the boundary conditions (25) and (26) replaced by:

$$\left(N_1(0) + \frac{\kappa}{2(1+\nu)} a_1(0) \right) \sum_{j=1}^n A_{1j}^{(1)} W_{1j} - \frac{\kappa}{2(1+\nu)} a_1(0) \Psi_{11} - l_1 K_{W1} W_{11} = 0 ;$$

$$K_{\Psi 1} \Psi_{11} - \frac{b_1(0)}{l_1} \sum_{j=1}^n A_{1j}^{(1)} \Psi_{1j} = 0$$

(34a,b)

$$\left(N_d(1) + \frac{\kappa}{2(1+\nu)} a_d(1) \right) \sum_{j=1}^n A_{nj}^{(1)} W_{dj} - \frac{\kappa}{2(1+\nu)} a_d(1) \Psi_{dn} - l_d K_{Wd} W_{dn} = 0 ;$$

$$K_{\Psi n} \Psi_{dn} - \frac{b_d(1)}{l_d} \sum_{j=1}^n A_{nj}^{(1)} \Psi_{dj} = 0$$

(35a,b)

The DQM linear equation system is used to determine the natural frequencies and mode shapes of the rotating beam.

The number of terms taken in the summations had been studied for many situations and the system has acceptable convergence by $n=21$ terms. (See Table 1)

4. Finite element method, MEF

An independent set of results for the natural frequencies, was also obtained by a finite element code. (Bambill et al., 2010). The finite element model employed in the analysis has 3000 beam elements of two nodes in the longitudinal direction (Rossi, 2007). See Table 2. This number of elements was proved to be enough with a convergence analysis.

The beam model also takes into account the shear deformation (Timoshenko beam’s theory) and the increase in bending stiffness induced by the centrifugal force.

The term $\rho I_k(\bar{x}_k) \bar{\eta}^2 \bar{\Psi}_k(\bar{x}_k)$ of equation (13.b) was not included in the finite element formulation. Probably for this reason some small differences between both sets of numerical results (DQM and FEM) begin to appear when the rotational speed η increases.

5. Numerical results

In the following examples some calculations were performed over elliptical cross sections. ($\kappa = 0.886364$). Without loss of generality, one may choose to keep constant width $e_k=e$ and vary the height $h_k(x)$ in each segment of the beam. The area and the second moment of area

of the cross section of the beam will be $A_k(x) = \frac{\pi e h_k(x)}{4}$, $I_k(x) = \frac{\pi e h_k^3(x)}{64}$, and for this

particular situation there are:

$$a_k(x) = \frac{h_k(x)}{h_k(0)} ; b_k(x) = \left(\frac{h_k(x)}{h_k(0)} \right)^3$$

The following formula is proposed to a quadratic variation of the height in each segment of beam:

$$h_k(x) = c_{0k} + c_{1k} x + c_{2k} x^2$$

And the slope is the derivative of this function

$$h'_k(x) = \frac{dh_k(x)}{dx} = c_{1k} + 2c_{2k}x$$

where c_{0k} , c_{1k} and c_{2k} are constants, which are defined by the heights and slopes at both ends of each segment k . The heights and slopes at each end are identified with the subscript A for $x=0$: h_{Ak} ; h'_{Ak} and with the subscript B for $x=1$: h_{Bk} ; h'_{Bk} .

If the segment of the beam shows a linear variation of height, $c_{2k} = 0$ and

$$h_{Ak} = c_{0k}; h_{Bk} = c_{0k} + c_{1k}; h'_{Ak} = h'_{Bk} = c_{1k}$$

As it can be seen in Table 1, the frequency coefficients calculated by the Differential Quadrature Method, DQM, using a summation with $n \geq 19$ ($i = 1, 2, 3, \dots, n$) points, show none significant improvement.

| n | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 |
|----|------------|------------|------------|------------|------------|
| 5 | 15.6861 | 29.2939 | 49.1602 | 63.9792 | 112.610 |
| 7 | 15.1981 | 28.9907 | 46.9070 | 64.9219 | 88.8670 |
| 9 | 14.9057 | 29.5079 | 47.4960 | 64.7054 | 87.4079 |
| 11 | 14.8340 | 29.6332 | 47.6579 | 64.7247 | 87.6724 |
| 13 | 14.8281 | 29.6467 | 47.6811 | 64.7310 | 87.7047 |
| 15 | 14.8291 | 29.6464 | 47.6820 | 64.7319 | 87.7079 |
| 17 | 14.8295 | 29.6460 | 47.6816 | 64.7320 | 87.7080 |
| 19 | 14.8296 | 29.6459 | 47.6815 | 64.7320 | 87.7080 |
| 21 | 14.8296 | 29.6459 | 47.6815 | 64.7320 | 87.7080 |

Table 1. Convergence analysis of the DQM, for a two-span rotating Timoshenko beam elastically restrained at both ends, with a quadratic variation of height.

The frequency coefficients in Table 1, correspond to a beam of two segments, rotating at speed $\eta = 10$, whose characteristics are: elliptical cross section; $\nu = 0.3$; $\kappa = 0.886364$; $R_1 = 0$; $l_1 / L = l_2 / L = 1 / 2$; $s_1 = \sqrt{300}$; $h_{B1} / h_{A1} = 1 / 2$; $h'_{B1} = 0$; $h_{A2} / h_{B1} = 1 / 2$; $h_{B2} / h_{A2} = 1 / 2$; $h'_{A2} = 0$; $K_{W1} = 10$; $K_{\psi 1} = 5$; $K_{Wd} = 0.1$; $K_{\psi d} = 1$.

In Table 2 the values obtained for the natural frequency coefficients using the finite element method are presented for $\eta = \sqrt{\rho A_0 / EI_0} L^2 \bar{\eta} = 0$ and $\eta = 10$. The number of elements is increased from 10 to 3000.

The model of the rotating beam of Table 2 has the following characteristics: one segment; rectangular cross section; $\nu = 0.3$; $\kappa = 10(1 + \nu) / (12 + 11\nu) = 0.849673$; $R_1 = 0$; $s_1 = \sqrt{300}$; $h_B / h_A = 1 / 4$; $h'_B = 0$; $K_{W1} \rightarrow \infty$; $K_{\psi 1} \rightarrow \infty$; $K_{Wd} = 0$; $K_{\psi d} = 0$.

In the first examples it is assumed a perfect clamped condition at the axis of rotation, given by: $K_{W1} \rightarrow \infty$ and $K_{\psi 1} \rightarrow \infty$. (Tables 3, 4 and 5).

Table 3 presents the effect of the rotational speed parameter η on the natural frequency coefficients of a rotating cantilever beam of one segment, ($K_{W1} \rightarrow \infty$; $K_{\psi 1} \rightarrow \infty$; $K_{Wd} = 0$; $K_{\psi d} = 0$). The results correspond to a linear variation of height and a comparison is made with

(Barnejee, 2006) when Banerjee’s parameter is $n=1$. As it can be observed the agreement is excellent.

| $\eta = 0$ | | | | | |
|--------------------|------------|------------|------------|------------|------------|
| Number of elements | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 |
| 10 | 3.38628165 | 11.7689336 | 26.5951854 | 46.6658427 | 71.0448001 |
| 100 | 3.37398143 | 11.7248502 | 26.4438604 | 46.1408176 | 69.5136708 |
| 1000 | 3.37385398 | 11.7243988 | 26.4423706 | 46.1357196 | 69.4986357 |
| 2000 | 3.37385302 | 11.7243954 | 26.4423593 | 46.1356810 | 69.4985219 |
| 3000 | 3.37385284 | 11.7243946 | 26.4423572 | 46.1356739 | 69.4985008 |
| $\eta = 10$ | | | | | |
| 10 | 11.6074237 | 25.8805102 | 44.0407905 | 66.3753084 | 92.6859627 |
| 100 | 11.6098042 | 25.7094320 | 43.5638284 | 65.4674874 | 90.8491237 |
| 1000 | 11.6098077 | 25.7074626 | 43.5585908 | 65.4579769 | 90.8301746 |
| 2000 | 11.6098078 | 25.7074476 | 43.5585511 | 65.4579049 | 90.8300310 |
| 3000 | 11.6098078 | 25.7074448 | 43.5585437 | 65.4578915 | 90.8300044 |

Table 2. Convergence analysis of the frequency coefficients $\Omega_i = \sqrt{\rho A_0 / EI_0} L^2 \omega_i$ using MEF.

| η | | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 |
|--------|-----------------|------------|------------|------------|------------|------------|
| 0 | DQM | 3.82377 | 18.3171 | 47.2638 | 90.4468 | 147.992 |
| | (Barnejee,2006) | 3.82379 | 18.3173 | 47.2648 | 90.4505 | 148.002 |
| 2 | DQM | 4.43680 | 18.9365 | 47.8706 | 91.0589 | 148.609 |
| | (Barnejee,2006) | 4.43680 | 18.9366 | 47.8717 | 91.0625 | 148.619 |
| 4 | DQM | 5.87874 | 20.6850 | 49.6446 | 92.8693 | 150.444 |
| | (Barnejee,2006) | 5.87877 | 20.6851 | 49.6456 | 92.8730 | 150.454 |
| 6 | DQM | 7.65512 | 23.3091 | 52.4622 | 95.8054 | 153.450 |
| | (Barnejee,2006) | 7.65514 | 23.3093 | 52.4632 | 95.8090 | 153.460 |
| 8 | DQM | 9.55392 | 26.5435 | 56.1584 | 99.7601 | 157.555 |
| | (Barnejee,2006) | 9.55396 | 26.5437 | 56.1595 | 99.7638 | 157.564 |
| 10 | DQM | 11.5015 | 30.1825 | 60.5628 | 104.608 | 162.668 |
| | (Barnejee,2006) | 11.5015 | 30.1827 | 60.5639 | 104.612 | 162.677 |

Table 3. Frequency coefficients $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a one-span beam, $l_1 / L = 1$; $s_1 = \sqrt{1000}$; $h_B / h_A = 1 / 2$; $K_{W1} \rightarrow \infty$; $K_{\psi 1} \rightarrow \infty$; $K_{Wd} = 0$; $K_{\psi d} = 0$.

All the calculations performed for the following Tables and Graphics used $R_1 = 0$; and $\nu = 0.30$; $\kappa = 0.886364$ (elliptical cross section).

The DQM results are determined using $n = 21$ in each segment of the beam, and the MEF results were obtained with 3000 elements.

The beam considered in Table 4 has one segment and is elastically restrained at its outer end. The parameter of rotation speed η is taken equal to 10. The Table presents the frequency coefficients for the first five mode shapes which correspond to different sets of elastically boundary conditions given by the spring constant parameters K_{Wd} and $K_{\psi d}$. The other details of the beam are specified in the legend of the table.

The beam model considered in Table 5 has two segments of equal length and similar conditions and parameters as Table 4.

| $K_{\psi d}$ | K_{Wd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 11.2148 | 27.6174 | 50.0089 | 77.5866 | 108.472 | |
| | | FEM | 11.2375 | 27.6743 | 50.0711 | 77.6432 | 108.523 | |
| | 0.1 | DQM | 15.4254 | 32.5178 | 52.8516 | 79.0733 | 109.357 | |
| | | FEM | 15.4438 | 32.5548 | 52.9087 | 79.1298 | 109.408 | |
| | 1 | DQM | 18.0157 | 40.3494 | 65.6538 | 92.2848 | 119.836 | |
| | | FEM | 18.0465 | 40.3841 | 65.6882 | 92.3208 | 119.877 | |
| | 10 | DQM | 18.3978 | 41.6361 | 69.0216 | 99.4111 | 131.859 | |
| | | FEM | 18.4315 | 41.6757 | 69.0611 | 99.4484 | 131.893 | |
| | $\rightarrow \infty$ | DQM | 18.4417 | 41.7750 | 69.3474 | 100.033 | 132.894 | |
| | | FEM | 18.4757 | 41.8151 | 69.3878 | 100.071 | 132.929 | |
| | 1 | 0 | DQM | 11.3941 | 29.3678 | 53.0174 | 81.0192 | 112.024 |
| | | | FEM | 11.4148 | 29.4104 | 53.0660 | 81.0662 | 112.068 |
| | | 0.1 | DQM | 15.6233 | 32.8965 | 55.0307 | 82.2247 | 112.825 |
| | | | FEM | 15.6400 | 32.9308 | 55.0763 | 82.2710 | 112.868 |
| 1 | | DQM | 19.2962 | 41.3980 | 65.9339 | 92.3365 | 120.822 | |
| | | FEM | 19.3219 | 41.4295 | 65.9674 | 92.3723 | 120.859 | |
| 10 | | DQM | 19.9179 | 43.3987 | 70.5662 | 100.622 | 132.723 | |
| | | FEM | 19.9463 | 43.4345 | 70.6034 | 100.658 | 132.756 | |
| $\rightarrow \infty$ | | DQM | 19.9899 | 43.6199 | 71.0558 | 101.509 | 134.136 | |
| | | FEM | 20.0187 | 43.6562 | 71.0937 | 101.546 | 134.170 | |
| 10 | | 0 | DQM | 11.4913 | 30.3954 | 55.1815 | 84.0260 | 115.628 |
| | | | FEM | 11.5115 | 30.4328 | 55.2229 | 84.0663 | 115.665 |
| | | 0.1 | DQM | 15.7621 | 33.1503 | 56.5688 | 84.8630 | 116.210 |
| | | | FEM | 15.7780 | 33.1835 | 56.6092 | 84.9031 | 116.247 |
| | 1 | DQM | 20.4765 | 42.5961 | 66.2635 | 92.3899 | 121.730 | |
| | | FEM | 20.4994 | 42.6248 | 66.2963 | 92.4255 | 121.765 | |
| | 10 | DQM | 21.3539 | 45.5548 | 72.8197 | 102.560 | 134.141 | |
| | | FEM | 21.3795 | 45.5875 | 72.8543 | 102.594 | 134.174 | |
| | $\rightarrow \infty$ | DQM | 21.4553 | 45.8807 | 73.5609 | 103.912 | 136.261 | |
| | | FEM | 21.4813 | 45.9139 | 73.5962 | 103.947 | 136.294 | |
| | $\rightarrow \infty$ | 0 | DQM | 11.5091 | 30.5860 | 55.6105 | 84.6706 | 116.454 |
| | | | FEM | 11.5291 | 30.6228 | 55.6510 | 84.7101 | 116.491 |
| | | 0.1 | DQM | 15.7905 | 33.2010 | 56.8768 | 85.4233 | 116.975 |
| | | | FEM | 15.8064 | 33.2340 | 56.9165 | 85.4625 | 117.012 |
| 1 | | DQM | 20.7557 | 42.9193 | 66.3549 | 92.4035 | 121.943 | |
| | | FEM | 20.7782 | 42.9475 | 66.3875 | 92.4392 | 121.978 | |
| 10 | | DQM | 21.6961 | 46.1510 | 73.5285 | 103.223 | 134.643 | |
| | | FEM | 21.7214 | 46.1832 | 73.5626 | 103.257 | 134.675 | |
| $\rightarrow \infty$ | | DQM | 21.8045 | 46.5039 | 74.3452 | 104.735 | 137.026 | |
| | | FEM | 21.8302 | 46.5368 | 74.3801 | 104.769 | 137.059 | |

Table 4. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a one-span rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $\nu = 0.3$; $s_1 = \sqrt{300}$; $h_B / h_A = 1 / 2$; $h'_B = 0$; $K_{W1} \rightarrow \infty$; $K_{\psi 1} \rightarrow \infty$; $\eta = 10$.

| $K_{\psi d}$ | K_{Vd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 11.8651 | 24.5717 | 40.8347 | 59.8775 | 81.1573 | |
| | | FEM | 11.8796 | 24.5914 | 40.8559 | 59.9110 | 81.1826 | |
| | 0.1 | DQM | 15.2667 | 30.3903 | 49.8409 | 67.9228 | 90.5546 | |
| | | FEM | 15.2858 | 30.4064 | 49.8638 | 67.9498 | 90.5743 | |
| | 1 | DQM | 15.6938 | 31.4585 | 52.2093 | 72.0062 | 98.7038 | |
| | | FEM | 15.7140 | 31.4754 | 52.2362 | 72.0330 | 98.7265 | |
| | 10 | DQM | 15.7412 | 31.5756 | 52.4458 | 72.4214 | 99.4803 | |
| | | FEM | 15.7616 | 31.5927 | 52.4732 | 72.4482 | 99.5038 | |
| | $\rightarrow \infty$ | DQM | 15.7466 | 31.5887 | 52.4718 | 72.4669 | 99.5627 | |
| | | FEM | 15.7669 | 31.6059 | 52.4993 | 72.4937 | 99.5862 | |
| | 1 | 0 | DQM | 11.9142 | 25.1342 | 42.8878 | 62.4877 | 85.6040 |
| | | | FEM | 11.9288 | 25.1532 | 42.9079 | 62.5196 | 85.6258 |
| 0.1 | | DQM | 16.2121 | 31.6526 | 50.5459 | 67.9979 | 90.8436 | |
| | | FEM | 16.2314 | 31.6672 | 50.5682 | 68.0245 | 90.8635 | |
| 1 | | DQM | 16.9952 | 33.8476 | 55.0090 | 75.3283 | 102.166 | |
| | | FEM | 17.0160 | 33.8634 | 55.0372 | 75.3520 | 102.190 | |
| 10 | | DQM | 17.0842 | 34.0961 | 55.4704 | 76.2542 | 103.723 | |
| | | FEM | 17.1052 | 34.1120 | 55.4993 | 76.2779 | 103.748 | |
| $\rightarrow \infty$ | | DQM | 17.0942 | 34.1238 | 55.5205 | 76.3541 | 103.882 | |
| | | FEM | 17.1152 | 34.1398 | 55.5496 | 76.3778 | 103.907 | |
| 10 | | 0 | DQM | 11.9157 | 25.1505 | 42.9498 | 62.5733 | 85.7690 |
| | | | FEM | 11.9302 | 25.1695 | 42.9699 | 62.6051 | 85.7907 |
| | 0.1 | DQM | 16.2528 | 31.7152 | 50.5831 | 68.0018 | 90.8571 | |
| | | FEM | 16.2721 | 31.7297 | 50.6053 | 68.0283 | 90.8770 | |
| | 1 | DQM | 17.0528 | 33.9729 | 55.1728 | 75.5622 | 102.430 | |
| | | FEM | 17.0737 | 33.9886 | 55.2011 | 75.5857 | 102.453 | |
| | 10 | DQM | 17.1437 | 34.2281 | 55.6450 | 76.5200 | 104.034 | |
| | | FEM | 17.1648 | 34.2440 | 55.6741 | 76.5435 | 104.059 | |
| | $\rightarrow \infty$ | DQM | 17.1539 | 34.2566 | 55.6962 | 76.6231 | 104.197 | |
| | | FEM | 17.1750 | 34.2725 | 55.7255 | 76.6466 | 104.222 | |
| | $\rightarrow \infty$ | 0 | DQM | 11.9158 | 25.1524 | 42.9569 | 62.5831 | 85.7880 |
| | | | FEM | 11.9304 | 25.1713 | 42.9770 | 62.6149 | 85.8097 |
| 0.1 | | DQM | 16.2575 | 31.7225 | 50.5875 | 68.0023 | 90.8587 | |
| | | FEM | 16.2768 | 31.7370 | 50.6097 | 68.0288 | 90.8786 | |
| 1 | | DQM | 17.0595 | 33.9876 | 55.1921 | 75.5901 | 102.461 | |
| | | FEM | 17.0804 | 34.0033 | 55.2205 | 75.6136 | 102.485 | |
| 10 | | DQM | 17.1506 | 34.2436 | 55.6656 | 76.5517 | 104.071 | |
| | | FEM | 17.1717 | 34.2595 | 55.6947 | 76.5752 | 104.096 | |
| $\rightarrow \infty$ | | DQM | 17.1609 | 34.2722 | 55.7169 | 76.6551 | 104.234 | |
| | | FEM | 17.1820 | 34.2881 | 55.7461 | 76.6786 | 104.259 | |

Table 5. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$ $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{W1} \rightarrow \infty$ $K_{\psi 1} \rightarrow \infty$, $\eta = 10$.

Next Tables, 6 to 10, correspond to beams of two segments, elastically restrained at both ends and any particular details are expressed in each legend.

| $K_{\psi d}$ | K_{Wd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 9.98841 | 21.2706 | 37.3110 | 54.9224 | 77.7336 | |
| | | FEM | 10.0246 | 21.3074 | 37.3506 | 54.9699 | 77.7658 | |
| | 0.1 | DQM | 12.4181 | 26.7466 | 45.4389 | 63.7717 | 87.4947 | |
| | | FEM | 12.4641 | 26.7810 | 45.4827 | 63.8054 | 87.5227 | |
| | 1 | DQM | 12.7051 | 27.6834 | 47.3154 | 67.8701 | 94.9448 | |
| | | FEM | 12.7526 | 27.7193 | 47.3634 | 67.9035 | 94.9788 | |
| | 10 | DQM | 12.7370 | 27.7869 | 47.5065 | 68.2901 | 95.6398 | |
| | | FEM | 12.7847 | 27.8230 | 47.5548 | 68.3236 | 95.6747 | |
| | $\rightarrow \infty$ | DQM | 12.7406 | 27.7985 | 47.5276 | 68.3362 | 95.7137 | |
| | | FEM | 12.7883 | 27.8347 | 47.5760 | 68.3697 | 95.7487 | |
| | 1 | 0 | DQM | 10.0086 | 21.6505 | 39.0007 | 57.4853 | 82.1930 |
| | | | FEM | 10.0451 | 21.6876 | 39.0412 | 57.5291 | 82.2227 |
| | | 0.1 | DQM | 13.1047 | 28.0775 | 46.2626 | 63.9615 | 87.6768 |
| | | | FEM | 13.1521 | 28.1101 | 46.3049 | 63.9941 | 87.7052 |
| 1 | | DQM | 13.6220 | 29.9765 | 49.8847 | 71.5084 | 98.2767 | |
| | | FEM | 13.6718 | 30.0120 | 49.9325 | 71.5384 | 98.3117 | |
| 10 | | DQM | 13.6812 | 30.1918 | 50.2664 | 72.4302 | 99.6646 | |
| | | FEM | 13.7312 | 30.2278 | 50.3147 | 72.4604 | 99.7012 | |
| $\rightarrow \infty$ | | DQM | 13.6879 | 30.2159 | 50.3082 | 72.5299 | 99.8074 | |
| | | FEM | 13.7379 | 30.2519 | 50.3566 | 72.5601 | 99.8441 | |
| 10 | | 0 | DQM | 10.0092 | 21.6615 | 39.0505 | 57.5689 | 82.3537 |
| | | | FEM | 10.0457 | 21.6987 | 39.0910 | 57.6127 | 82.3835 |
| | | 0.1 | DQM | 13.1336 | 28.1416 | 46.3059 | 63.9714 | 87.6854 |
| | | | FEM | 13.1811 | 28.1741 | 46.3481 | 64.0039 | 87.7138 |
| | 1 | DQM | 13.6618 | 30.0922 | 50.0321 | 71.7561 | 98.5256 | |
| | | FEM | 13.7116 | 30.1278 | 50.0799 | 71.7860 | 98.5607 | |
| | 10 | DQM | 13.7222 | 30.3131 | 50.4233 | 72.7081 | 99.9556 | |
| | | FEM | 13.7723 | 30.3491 | 50.4716 | 72.7381 | 99.9923 | |
| | $\rightarrow \infty$ | DQM | 13.7290 | 30.3379 | 50.4661 | 72.8107 | 100.102 | |
| | | FEM | 13.7792 | 30.3739 | 50.5145 | 72.8408 | 100.139 | |
| | $\rightarrow \infty$ | 0 | DQM | 10.0093 | 21.6628 | 39.0562 | 57.5786 | 82.3722 |
| | | | FEM | 10.0458 | 21.6999 | 39.0968 | 57.6224 | 82.4020 |
| | | 0.1 | DQM | 13.1369 | 28.1491 | 46.3110 | 63.9726 | 87.6864 |
| | | | FEM | 13.1844 | 28.1817 | 46.3532 | 64.0051 | 87.7149 |
| 1 | | DQM | 13.6664 | 30.1057 | 50.0495 | 71.7856 | 98.5555 | |
| | | FEM | 13.7163 | 30.1414 | 50.0973 | 71.8155 | 98.5906 | |
| 10 | | DQM | 13.7270 | 30.3273 | 50.4418 | 72.7411 | 99.9903 | |
| | | FEM | 13.7771 | 30.3634 | 50.4901 | 72.7711 | 100.027 | |
| $\rightarrow \infty$ | | DQM | 13.7338 | 30.3521 | 50.4847 | 72.8441 | 100.137 | |
| | | FEM | 13.7840 | 30.3882 | 50.5331 | 72.8741 | 100.174 | |

Table 6. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$ $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{W1} \rightarrow \infty$, $K_{\psi 1} = 0.1$, $\eta = 10$.

| $K_{\psi d}$ | K_{Vd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 11.3734 | 23.4059 | 39.2570 | 56.9363 | 78.4787 | |
| | | FEM | 11.3904 | 23.4273 | 39.2820 | 56.9721 | 78.5041 | |
| | 0.1 | DQM | 14.4551 | 28.9705 | 47.6260 | 65.1400 | 87.8678 | |
| | | FEM | 14.4773 | 28.9890 | 47.6542 | 65.1664 | 87.8894 | |
| | 1 | DQM | 14.8320 | 29.9641 | 49.6507 | 69.1005 | 95.0494 | |
| | | FEM | 14.8553 | 29.9837 | 49.6827 | 69.1264 | 95.0757 | |
| | 10 | DQM | 14.8738 | 30.0733 | 49.8536 | 69.5057 | 95.7100 | |
| | | FEM | 14.8972 | 30.0931 | 49.8861 | 69.5316 | 95.7371 | |
| | $\rightarrow \infty$ | DQM | 14.8785 | 30.0856 | 49.8761 | 69.5501 | 95.7801 | |
| | | FEM | 14.9019 | 30.1054 | 49.9086 | 69.5761 | 95.8073 | |
| | 1 | 0 | DQM | 11.4126 | 23.8883 | 41.1048 | 59.4030 | 82.7965 |
| | | | FEM | 11.4297 | 23.9092 | 41.1294 | 59.4361 | 82.8192 |
| 0.1 | | DQM | 15.3087 | 30.2367 | 48.3683 | 65.2804 | 88.0547 | |
| | | FEM | 15.3312 | 30.2539 | 48.3957 | 65.3061 | 88.0766 | |
| 1 | | DQM | 15.9925 | 32.2677 | 52.2109 | 72.5363 | 98.2033 | |
| | | FEM | 16.0166 | 32.2865 | 52.2436 | 72.5592 | 98.2306 | |
| 10 | | DQM | 16.0702 | 32.4974 | 52.6080 | 73.4315 | 99.5067 | |
| | | FEM | 16.0945 | 32.5165 | 52.6413 | 73.4545 | 99.5353 | |
| $\rightarrow \infty$ | | DQM | 16.0790 | 32.5231 | 52.6513 | 73.5282 | 99.6404 | |
| | | FEM | 16.1032 | 32.5422 | 52.6847 | 73.5513 | 99.6691 | |
| 10 | | 0 | DQM | 11.4138 | 23.9023 | 41.1598 | 59.4837 | 82.9526 |
| | | | FEM | 11.4309 | 23.9232 | 41.1844 | 59.5168 | 82.9753 |
| | 0.1 | DQM | 15.3450 | 30.2988 | 48.4074 | 65.2877 | 88.0636 | |
| | | FEM | 15.3676 | 30.3159 | 48.4348 | 65.3134 | 88.0854 | |
| | 1 | DQM | 16.0434 | 32.3866 | 52.3585 | 72.7730 | 98.4379 | |
| | | FEM | 16.0676 | 32.4055 | 52.3914 | 72.7958 | 98.4653 | |
| | 10 | DQM | 16.1228 | 32.6224 | 52.7651 | 73.6979 | 99.7796 | |
| | | FEM | 16.1471 | 32.6415 | 52.7985 | 73.7208 | 99.8083 | |
| | $\rightarrow \infty$ | DQM | 16.1317 | 32.6487 | 52.8094 | 73.7975 | 99.9164 | |
| | | FEM | 16.1560 | 32.6679 | 52.8428 | 73.8204 | 99.9453 | |
| | $\rightarrow \infty$ | 0 | DQM | 11.4139 | 23.9039 | 41.1661 | 59.4929 | 82.9706 |
| | | | FEM | 11.4310 | 23.9248 | 41.1908 | 59.5260 | 82.9933 |
| 0.1 | | DQM | 15.3493 | 30.3061 | 48.4120 | 65.2886 | 88.0646 | |
| | | FEM | 15.3718 | 30.3232 | 48.4394 | 65.3142 | 88.0865 | |
| 1 | | DQM | 16.0493 | 32.4005 | 52.3759 | 72.8012 | 98.4661 | |
| | | FEM | 16.0735 | 32.4194 | 52.4088 | 72.8240 | 98.4934 | |
| 10 | | DQM | 16.1289 | 32.6371 | 52.7836 | 73.7295 | 99.8122 | |
| | | FEM | 16.1532 | 32.6562 | 52.8170 | 73.7524 | 99.8409 | |
| $\rightarrow \infty$ | | DQM | 16.1378 | 32.6635 | 52.8280 | 73.8295 | 99.9493 | |
| | | FEM | 16.1622 | 32.6827 | 52.8615 | 73.8524 | 99.9782 | |

Table 7. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$, $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{W1} = 10$, $K_{\psi 1} = 10$, $\eta = 10$.

| $K_{\psi d}$ | K_{Vd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 11.0954 | 22.8658 | 38.6771 | 56.1987 | 78.0667 | |
| | | FEM | 11.1149 | 22.8898 | 38.7052 | 56.2367 | 78.0931 | |
| | 0.1 | DQM | 14.0189 | 28.3688 | 46.9190 | 64.5736 | 87.5306 | |
| | | FEM | 14.0444 | 28.3902 | 46.9506 | 64.6011 | 87.5532 | |
| | 1 | DQM | 14.3726 | 29.3408 | 48.8766 | 68.5581 | 94.6820 | |
| | | FEM | 14.3992 | 29.3634 | 48.9120 | 68.5850 | 94.7094 | |
| | 10 | DQM | 14.4118 | 29.4478 | 49.0737 | 68.9659 | 95.3401 | |
| | | FEM | 14.4386 | 29.4706 | 49.1095 | 68.9930 | 95.3684 | |
| | $\rightarrow \infty$ | DQM | 14.4162 | 29.4598 | 49.0954 | 69.0107 | 95.4100 | |
| | | FEM | 14.4430 | 29.4826 | 49.1314 | 69.0377 | 95.4383 | |
| | 1 | 0 | DQM | 11.1299 | 23.3178 | 40.4680 | 58.6771 | 82.4016 |
| | | | FEM | 11.1495 | 23.3414 | 40.4960 | 58.7122 | 82.4254 |
| 0.1 | | DQM | 14.8296 | 29.6459 | 47.6815 | 64.7320 | 87.7080 | |
| | | FEM | 14.8556 | 29.6659 | 47.7122 | 64.7587 | 87.7309 | |
| 1 | | DQM | 15.4691 | 31.6287 | 51.4150 | 72.0511 | 97.8483 | |
| | | FEM | 15.4967 | 31.6508 | 51.4509 | 72.0751 | 97.8766 | |
| 10 | | DQM | 15.5418 | 31.8531 | 51.8028 | 72.9497 | 99.1495 | |
| | | FEM | 15.5696 | 31.8754 | 51.8392 | 72.9737 | 99.1791 | |
| $\rightarrow \infty$ | | DQM | 15.5499 | 31.8782 | 51.8452 | 73.0468 | 99.2832 | |
| | | FEM | 15.5778 | 31.9005 | 51.8817 | 73.0708 | 99.3129 | |
| 10 | | 0 | DQM | 11.1309 | 23.3309 | 40.5211 | 58.7582 | 82.5578 |
| | | | FEM | 11.1505 | 23.3545 | 40.5491 | 58.7932 | 82.5816 |
| | 0.1 | DQM | 14.8640 | 29.7082 | 47.7217 | 64.7403 | 87.7163 | |
| | | FEM | 14.8900 | 29.7282 | 47.7523 | 64.7669 | 87.7393 | |
| | 1 | DQM | 15.5170 | 31.7461 | 51.5610 | 72.2905 | 98.0837 | |
| | | FEM | 15.5447 | 31.7682 | 51.5970 | 72.3143 | 98.1121 | |
| | 10 | DQM | 15.5912 | 31.9764 | 51.9582 | 73.2187 | 99.4233 | |
| | | FEM | 15.6191 | 31.9987 | 51.9947 | 73.2426 | 99.4531 | |
| | $\rightarrow \infty$ | DQM | 15.5996 | 32.0021 | 52.0016 | 73.3186 | 99.5601 | |
| | | FEM | 15.6275 | 32.0245 | 52.0381 | 73.3426 | 99.5900 | |
| | $\rightarrow \infty$ | 0 | DQM | 11.1310 | 23.3324 | 40.5272 | 58.7675 | 82.5758 |
| | | | FEM | 11.1506 | 23.3560 | 40.5552 | 58.8025 | 82.5996 |
| 0.1 | | DQM | 14.8680 | 29.7155 | 47.7264 | 64.7412 | 87.7173 | |
| | | FEM | 14.8940 | 29.7355 | 47.7570 | 64.7679 | 87.7402 | |
| 1 | | DQM | 15.5225 | 31.7598 | 51.5783 | 72.3191 | 98.1119 | |
| | | FEM | 15.5503 | 31.7819 | 51.6143 | 72.3428 | 98.1403 | |
| 10 | | DQM | 15.5970 | 31.9908 | 51.9765 | 73.2506 | 99.4560 | |
| | | FEM | 15.6249 | 32.0132 | 52.0131 | 73.2745 | 99.4857 | |
| $\rightarrow \infty$ | | DQM | 15.6053 | 32.0166 | 52.0200 | 73.3510 | 99.5931 | |
| | | FEM | 15.6333 | 32.0391 | 52.0566 | 73.3749 | 99.6230 | |

Table 8. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$, $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{W1} = 10$, $K_{\psi 1} = 5$, $\eta = 10$.

| $K_{\psi d}$ | K_{Vd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 10.3650 | 21.7083 | 37.5771 | 54.9536 | 77.3897 | |
| | | FEM | 10.3937 | 21.7398 | 37.6121 | 54.9961 | 77.4183 | |
| | 0.1 | DQM | 12.9366 | 27.1514 | 45.6367 | 63.6520 | 86.9652 | |
| | | FEM | 12.9736 | 27.1805 | 45.6754 | 63.6816 | 86.9900 | |
| | 1 | DQM | 13.2417 | 28.0899 | 47.4960 | 67.6756 | 94.0576 | |
| | | FEM | 13.2800 | 28.1204 | 47.5385 | 67.7048 | 94.0875 | |
| | 10 | DQM | 13.2755 | 28.1934 | 47.6848 | 68.0876 | 94.7110 | |
| | | FEM | 13.3141 | 28.2242 | 47.7276 | 68.1169 | 94.7416 | |
| | $\rightarrow \infty$ | DQM | 13.2794 | 28.2050 | 47.7057 | 68.1329 | 94.7804 | |
| | | FEM | 13.3179 | 28.2358 | 47.7485 | 68.1621 | 94.8111 | |
| | 1 | 0 | DQM | 10.3892 | 22.1044 | 39.2705 | 57.4666 | 81.7463 |
| | | | FEM | 10.4182 | 22.1360 | 39.3061 | 57.5054 | 81.7724 |
| 0.1 | | DQM | 13.6565 | 28.4599 | 46.4416 | 63.8427 | 87.1268 | |
| | | FEM | 13.6946 | 28.4875 | 46.4789 | 63.8714 | 87.1519 | |
| 1 | | DQM | 14.2060 | 30.3646 | 50.0205 | 71.2606 | 97.2421 | |
| | | FEM | 14.2462 | 30.3947 | 50.0627 | 71.2866 | 97.2727 | |
| 10 | | DQM | 14.2687 | 30.5803 | 50.3961 | 72.1634 | 98.5382 | |
| | | FEM | 14.3091 | 30.6108 | 50.4388 | 72.1896 | 98.5700 | |
| $\rightarrow \infty$ | | DQM | 14.2757 | 30.6044 | 50.4372 | 72.2610 | 98.6715 | |
| | | FEM | 14.3161 | 30.6350 | 50.4800 | 72.2872 | 98.7035 | |
| 10 | | 0 | DQM | 10.3900 | 22.1160 | 39.3204 | 57.5486 | 81.9025 |
| | | | FEM | 10.4190 | 22.1475 | 39.3560 | 57.5874 | 81.9286 |
| | 0.1 | DQM | 13.6869 | 28.5231 | 46.4839 | 63.8527 | 87.1344 | |
| | | FEM | 13.7250 | 28.5506 | 46.5212 | 63.8813 | 87.1595 | |
| | 1 | DQM | 14.2479 | 30.4798 | 50.1652 | 71.5043 | 97.4786 | |
| | | FEM | 14.2881 | 30.5099 | 50.2075 | 71.5301 | 97.5092 | |
| | 10 | DQM | 14.3119 | 30.7011 | 50.5501 | 72.4364 | 98.8132 | |
| | | FEM | 14.3524 | 30.7317 | 50.5928 | 72.4624 | 98.8452 | |
| | $\rightarrow \infty$ | DQM | 14.3191 | 30.7259 | 50.5922 | 72.5368 | 98.9498 | |
| | | FEM | 14.3596 | 30.7565 | 50.6350 | 72.5629 | 98.9819 | |
| | $\rightarrow \infty$ | 0 | DQM | 10.3901 | 22.1173 | 39.3262 | 57.5580 | 81.9205 |
| | | | FEM | 10.4190 | 22.1488 | 39.3618 | 57.5968 | 81.9466 |
| 0.1 | | DQM | 13.6904 | 28.5305 | 46.4889 | 63.8539 | 87.1353 | |
| | | FEM | 13.7286 | 28.5580 | 46.5261 | 63.8825 | 87.1604 | |
| 1 | | DQM | 14.2527 | 30.4933 | 50.1823 | 71.5333 | 97.5069 | |
| | | FEM | 14.2930 | 30.5235 | 50.2245 | 71.5591 | 97.5375 | |
| 10 | | DQM | 14.3169 | 30.7153 | 50.5682 | 72.4688 | 98.8460 | |
| | | FEM | 14.3574 | 30.7458 | 50.6110 | 72.4948 | 98.8780 | |
| $\rightarrow \infty$ | | DQM | 14.3241 | 30.7401 | 50.6105 | 72.5696 | 98.9829 | |
| | | FEM | 14.3646 | 30.7707 | 50.6532 | 72.5957 | 99.0150 | |

Table 9. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$, $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{V1} = 10$, $K_{\psi 1} = 1$, $\eta = 10$.

| $K_{\psi d}$ | K_{Vd} | Method | Ω_1 | Ω_2 | Ω_3 | Ω_4 | Ω_5 | |
|----------------------|----------------------|--------|------------|------------|------------|------------|------------|---------|
| 0 | 0 | DQM | 9.94295 | 21.1789 | 37.1287 | 54.4936 | 77.1422 | |
| | | FEM | 9.97877 | 21.2145 | 37.1668 | 54.5377 | 77.1717 | |
| | 0.1 | DQM | 12.3497 | 26.6244 | 45.1340 | 63.3202 | 86.7544 | |
| | | FEM | 12.3950 | 26.6574 | 45.1757 | 63.3506 | 86.7801 | |
| | 1 | DQM | 12.6335 | 27.5522 | 46.9619 | 67.3571 | 93.8220 | |
| | | FEM | 12.6802 | 27.5867 | 47.0072 | 67.3872 | 93.8529 | |
| | 10 | DQM | 12.6651 | 27.6547 | 47.1481 | 67.7705 | 94.4732 | |
| | | FEM | 12.7119 | 27.6894 | 47.1937 | 67.8007 | 94.5049 | |
| | $\rightarrow \infty$ | DQM | 12.6686 | 27.6662 | 47.1687 | 67.8159 | 94.5425 | |
| | | FEM | 12.7155 | 27.7009 | 47.2144 | 67.8461 | 94.5743 | |
| | 1 | 0 | DQM | 9.96265 | 21.5532 | 38.7860 | 57.0231 | 81.5044 |
| | | | FEM | 9.99874 | 21.5890 | 38.8248 | 57.0634 | 81.5315 |
| 0.1 | | DQM | 13.0297 | 27.9504 | 45.9575 | 63.5235 | 86.9102 | |
| | | FEM | 13.0763 | 27.9817 | 45.9976 | 63.5530 | 86.9362 | |
| 1 | | DQM | 13.5409 | 29.8282 | 49.4887 | 70.9743 | 97.0122 | |
| | | FEM | 13.5897 | 29.8623 | 49.5335 | 71.0012 | 97.0437 | |
| 10 | | DQM | 13.5994 | 30.0410 | 49.8608 | 71.8782 | 98.3059 | |
| | | FEM | 13.6484 | 30.0754 | 49.9060 | 71.9052 | 98.3387 | |
| $\rightarrow \infty$ | | DQM | 13.6060 | 30.0648 | 49.9016 | 71.9759 | 98.4391 | |
| | | FEM | 13.6550 | 30.0993 | 49.9469 | 72.0030 | 98.4720 | |
| 10 | | 0 | DQM | 9.96324 | 21.5641 | 38.8347 | 57.1056 | 81.6606 |
| | | | FEM | 9.99933 | 21.5999 | 38.8736 | 57.1459 | 81.6876 |
| | 0.1 | DQM | 13.0583 | 28.0142 | 46.0008 | 63.5342 | 86.9175 | |
| | | FEM | 13.1049 | 28.0454 | 46.0408 | 63.5635 | 86.9436 | |
| | 1 | DQM | 13.5802 | 29.9428 | 49.6333 | 71.2194 | 97.2490 | |
| | | FEM | 13.6291 | 29.9769 | 49.6781 | 71.2462 | 97.2806 | |
| | 10 | DQM | 13.6399 | 30.1611 | 50.0148 | 72.1525 | 98.5813 | |
| | | FEM | 13.6890 | 30.1956 | 50.0600 | 72.1794 | 98.6142 | |
| | $\rightarrow \infty$ | DQM | 13.6467 | 30.1856 | 50.0566 | 72.2531 | 98.7177 | |
| | | FEM | 13.6958 | 30.2201 | 50.1018 | 72.2800 | 98.7508 | |
| | $\rightarrow \infty$ | 0 | DQM | 9.96331 | 21.5653 | 38.8403 | 57.1151 | 81.6785 |
| | | | FEM | 9.99940 | 21.6011 | 38.8792 | 57.1553 | 81.7056 |
| 0.1 | | DQM | 13.0616 | 28.0216 | 46.0058 | 63.5354 | 86.9184 | |
| | | FEM | 13.1082 | 28.0529 | 46.0459 | 63.5648 | 86.9444 | |
| 1 | | DQM | 13.5848 | 29.9563 | 49.6504 | 71.2486 | 97.2773 | |
| | | FEM | 13.6337 | 29.9904 | 49.6952 | 71.2753 | 97.3090 | |
| 10 | | DQM | 13.6447 | 30.1752 | 50.0329 | 72.1851 | 98.6141 | |
| | | FEM | 13.6938 | 30.2097 | 50.0781 | 72.2120 | 98.6471 | |
| $\rightarrow \infty$ | | DQM | 13.6514 | 30.1997 | 50.0748 | 72.2860 | 98.7509 | |
| | | FEM | 13.7006 | 30.2342 | 50.1201 | 72.3129 | 98.7840 | |

Table 10. First natural frequencies $\Omega_i = \sqrt{\rho A_1(0) / EI_1(0)} L^2 \omega_i$ for a two-span elastically restrained rotating Timoshenko beam, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = 1 / 2$, $l_2 / L = 1 / 2$, $h_{B1} / h_{A1} = 1 / 2$, $h'_{B1} = 0$, $h_{A2} / h_{B1} = 1 / 2$, $h_{B2} / h_{A2} = 1 / 2$, $h'_{A2} = 0$, $K_{V1} = 10$, $K_{\psi 1} = 0.1$, $\eta = 10$.

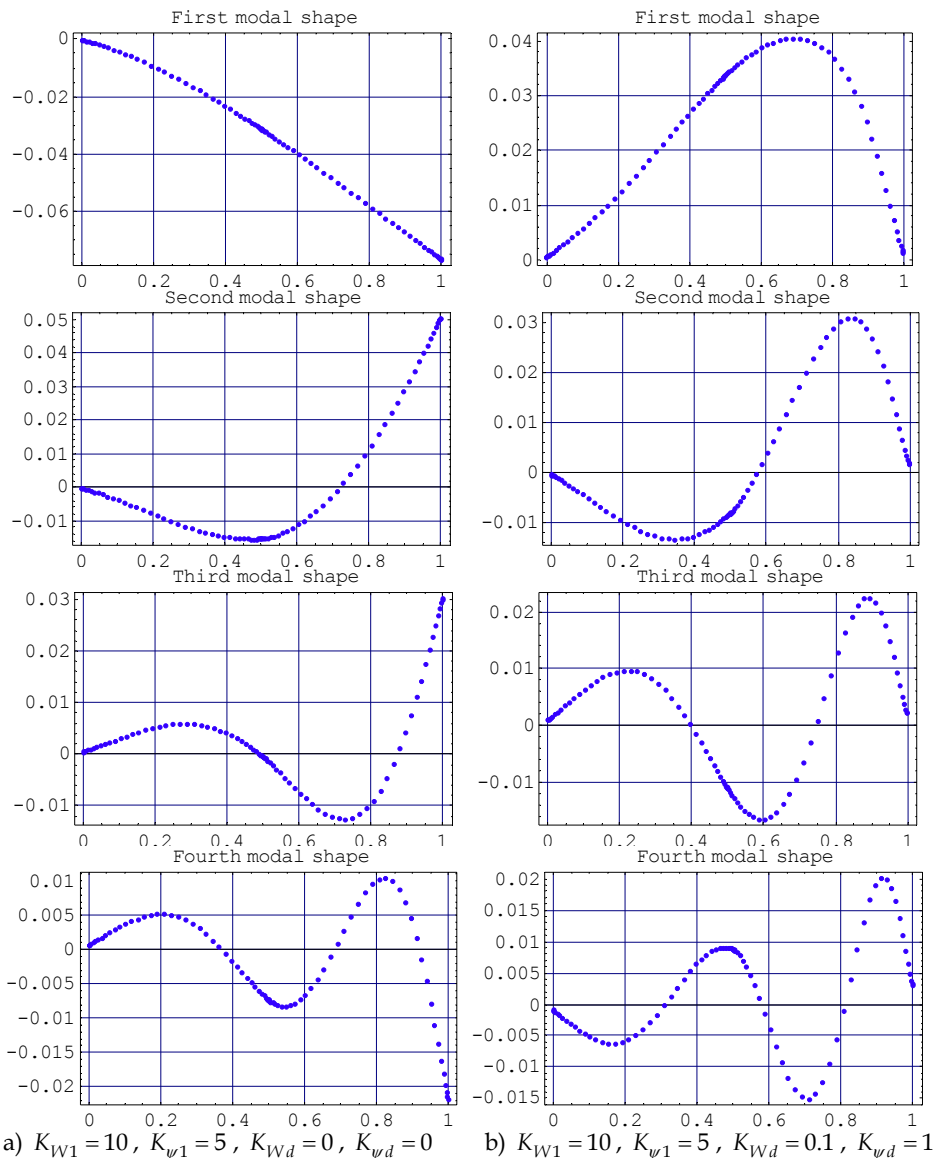


Fig. 4. Natural frequencies mode shapes for a two-span elastically restrained rotating Timoshenko beams, with elliptical cross section and quadratic height variation along the axis. $l_1 / L = l_2 / L = 1 / 2 ; h_{B1} / h_{A1} = 1 / 2 ; h'_{B1} = 0 ; h_{A2} / h_{B1} = 1 / 2 ; h_{B2} / h_{A2} = 1 / 2 ; h'_{A2} = 0 ; \eta = 10$

Figure 4 shows the first four natural frequency mode shapes for beams, with two different kinds of boundary conditions: a) corresponds to $K_{W1} = 10, K_{\psi 1} = 5, K_{Wd} = 0, K_{\psi d} = 0$, while b) corresponds to $K_{W1} = 10, K_{\psi 1} = 5, K_{Wd} = 0.1, K_{\psi d} = 1$.

The next Figures, 5 and 6, present the variation of the fundament frequency parameter Ω_1 with the variation of the non-dimensional rotational speed η and the spring constant $K\psi_1$.

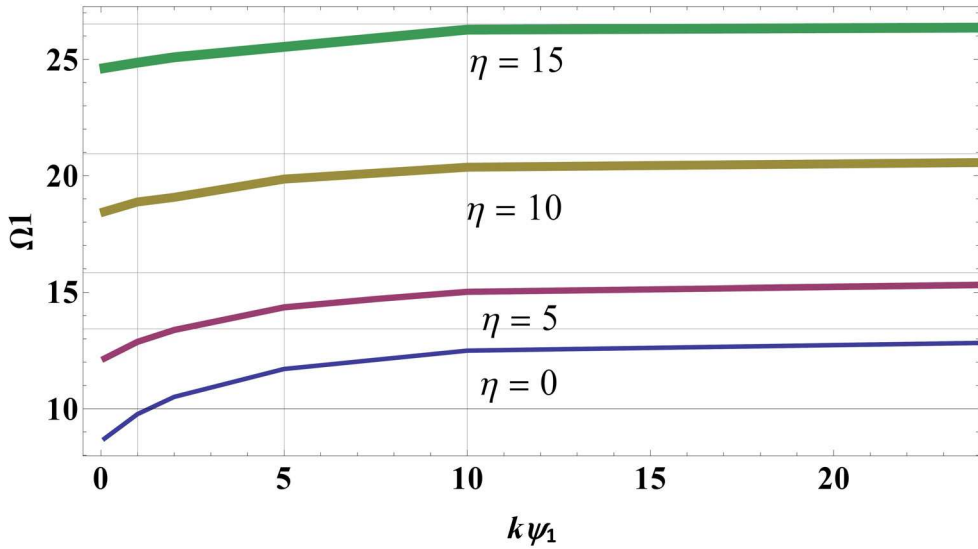


Fig. 5. The fundamental frequency coefficient Ω_1 of a one-span elastically restrained rotating Timoshenko beam versus the spring constant parameter of the rotational spring $K\psi_1$, for different rotational speed parameters η . $K_{w1}=10$; $K_{wd}=1$; $K_{\psi d}=10$

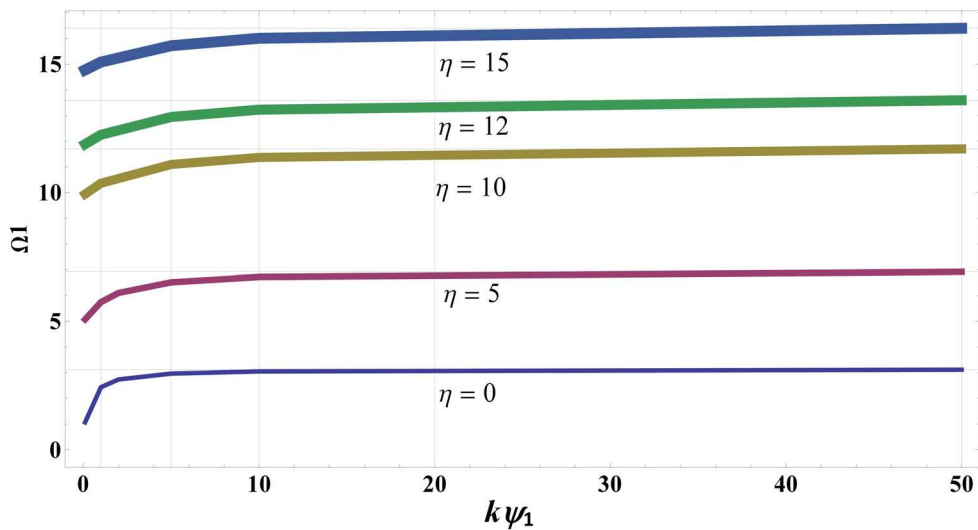


Fig. 6. The fundamental frequency coefficient Ω_1 of a two-span elastically restrained rotating Timoshenko beam versus the spring constant parameter of the rotational spring $K\psi_1$, for different rotational speed parameters η . $K_{w1}=10$; $K_{wd}=0$; $K_{\psi d}=0$

6. Conclusion

The differential quadrature method proves to be very efficient to obtain frequencies and mode shapes of natural vibration, for the rotating Timoshenko beam model.

The versatility of the proposed beam model (variable cross section, step change in cross section, elastic restraints at both ends) allows to solve a large number of individual cases.

Something interesting to point out is that because the method directly solves two ordinary differential equations, additional restrictions are not generated. This does not happen in other methodologies, such as the dynamic stiffness method (Banerjee, 2000, 2001).

As a matter of fact, the differential quadrature method has the same advantage as the finite element method and it needs less computer memory requirements than the FEM.

In particular the present results show that the frequency coefficients vary more significantly when the translational spring stiffness changes at the end of the beam farthest from the axis of rotation K_{qd} .

7. Appendix A

As Shu presents in his book (Shu, 2000), the differential quadrature method, DQM, is a numerical technique for solving differential equations.

In order to obtain the DQM analog equations to the governing equations of the rotating beam and its boundary conditions, the beam domain is discretized in a grid of points using the Chebyshev - Gauss - Lobato expression, (Shu & Chen, 1999):

$$x_i = \frac{1 - \cos[(i - 1)\pi / (n - 1)]}{2} ; i = 1, 2, \dots, n$$

where n is the number of discrete points or nodes and x_i is the coordinate of node i .

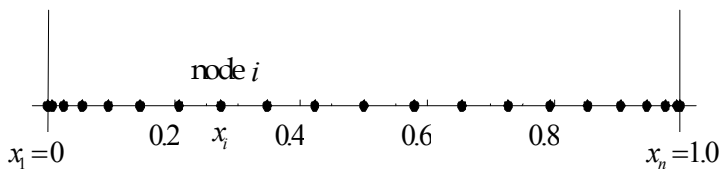


Fig. A1. Grid of n points

The weighting coefficients $A_{ij}^{(1)}$ and $A_{ij}^{(2)}$, which appeared in the linear algebraic equations of quadrature (28-35), were determined using the explicit expressions cited by (Bert & Malik, 1996).

The coefficients $A_{ij}^{(1)}$ correspond to first order derivatives and can be arranged in a square matrix of order n .

The matrix elements $A_{ij}^{(1)}$ with $i \neq j$, are determined by:

$$A_{ij}^{(1)} = \frac{\prod(x_i)}{(x_i - x_j) \prod(x_j)}$$

where

$$\prod(x_i) = \prod_{v=1, v \neq i}^n (x_i - x_v); \quad \prod(x_j) = \prod_{v=1, v \neq i}^n (x_j - x_v);$$

The coefficients $A_{ij}^{(1)}$ with $i = j$, will tend to infinity and need to be calculated in another way.

The coefficients $A_{ij}^{(2)}$ correspond to second-order derivatives and are obtained from

$$A_{ij}^{(2)} = 2 \left[A_{ii}^{(1)} * A_{ij}^{(1)} - \frac{A_{ij}^{(1)}}{x_i - x_j} \right]$$

with $i \neq j$ and $i, j = 1, 2, 3, \dots, n$.

Because the sum of the weighting coefficients of a row of the matrix is zero, it is easy to calculate the diagonal terms of derivatives of any order q , using the following expression:

$$A_{ii}^{(q)} = - \sum_{j=1, j \neq i}^n A_{ij}^{(q)}$$

And the equations for q equal to 1 and 2, corresponding to first and second order derivatives, are:

$$A_{ii}^{(1)} = - \sum_{j=1, j \neq i}^n A_{ij}^{(1)}; \quad A_{ii}^{(2)} = - \sum_{j=1, j \neq i}^n A_{ij}^{(2)}$$

8. Acknowledgment

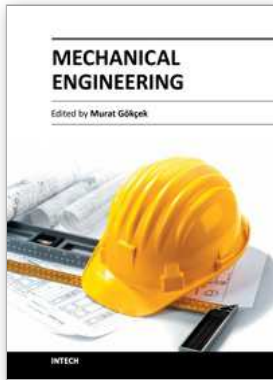
The authors gratefully acknowledge the support of the Universidad Nacional del Sur (UNS) and the Consejo Nacional de Investigaciones Científicas y Técnicas (CONICET), Argentina.

9. References

- Bambill, D.V.; Felix, D.H. & Rossi, R. E. (2010). Vibration analysis of rotating Timoshenko beams by means of the differential quadrature method. *Structural Engineering and Mechanics*, Vol. 34, No. 2, pp. 231-245, ISSN 12254568
- Banerjee, J. (2000). Free vibration of centrifugally stiffened uniform and tapered beams using the dynamic stiffness method. *Journal of Sound and Vibration*, Vol.233, No.5, pp. 857-875, ISSN 0022-460X
- Banerjee, J. (2001). Dynamic stiffness formulation and free vibration analysis of centrifugally stiffened Timoshenko beam. *Journal of Sound and Vibration*, Vol.247, pp. 97-115, ISSN 0022-460X

- Banerjee, J.; Su, H. & Jackson, D. (2006). Free vibration of rotating tapered beams using the dynamic stiffness method. *Journal of Sound and Vibration*, Vol. 298, pp. 1034-1054, ISSN 0022-460X
- Bellman, R. & Casti, J. (1971). Differential quadrature and long-term Integration. *J. Math. Anal.* Vol.34, pp. 235-238, ISSN 0022-247X
- Bellman, R.E. & Roth, R.S. (1986). *Methods in approximation: techniques for mathematical modelling*, Editorial D. Reidel Publishing Company, ISBN 9-027-72188-2, Dordrecht, Holland
- Bert, C. & Malik, M. (1996). Differential quadrature method in computational mechanics: A review. *Applied Mechanics Review* Vol.49, pp. 1-28, ISSN 0008-6900
- Choi, S.; Wu J. & Chou Y. (2000). Dynamic analysis of a spinning Timoshenko beam by the differential quadrature method. *American Institute of Aeronautics and Astronautics* Vol.38, pp. 51-856, ISSN 0001-1452
- Felix, D.H.; Rossi, R. E. & Bambill, D. V. (2008). Vibraciones transversales por el método de cuadratura diferencial de una viga Timoshenko rotante, escalonada y elásticamente vinculada, *Mecánica Computacional* Vol. XXVII, pp.1957-1973, ISBN 1666-6070
- Felix, D. H.; Bambill, D. V. & Rossi, R. E. (2009). Análisis de vibración libre de una viga Timoshenko escalonada, centrifugamente rigidizada, mediante el método de cuadratura diferencial, *Revista Internacional de Métodos Numéricos para Cálculo y Diseño en Ingeniería*. Vol. 25, No. 2, pp. 111-132, ISSN 0213-1315
- Ganesh, R and Ganguli, R. (2011). Physics based basis function for vibration analysis of high speed rotating beams. *Structural Engineering and Mechanics*, Vol.39, No.1, pp. 21-46, ISSN 1225-4568
- Gunda, J. B. & Ganguli R. (2008). New rational interpolation functions for finite element analysis of rotating beams. *International Journal of Mechanical Sciences*; Vol. 50, pp. 578-588, ISSN 0020-7403
- Gunda, J.B.; Singh, A.P.; Chhabra, P.S. & Ganguli, R. (2007). Free vibration analysis of rotating tapered blades using Fourier-p superelement, *Structural Engineering and Mechanics*, Vol.27, No.2, pp. 243-257, ISSN 1225-4568
- Kumar A. & Ganguli R. (2009). Rotating Beams and Nonrotating Beams with Shared Eigenpair, *Journal of Applied Mechanics*. Vol.76. No.5, pp. 1-14, ISSN: 0021-8936
- Hodges, D. H. & Rutkowski, M. J. (1981). Free vibration analysis of rotating beams by a variable order finite method, *American Institute of Aeronautics and Astronautics Journal*. Vol.19, No.11, pp. 1459-1466
- Lin, S. C. & Hsiao, K. M. (2001). Vibration analysis of a rotating Timoshenko beam. *Journal of Sound and Vibration*. Vol. 240, pp. 303-322.
- Liu, G. R. & Wu, T. Y. (2001). Vibration analysis of beams using the generalized differential quadrature rule and domain decomposition. *Journal of Sound and Vibration*. Vol.246, pp.461-481, ISSN 0022-460X
- Naguleswaran, S. (2004). Transverse vibration and stability of an Euler-Bernoulli beam with step change in cross-section and in axial force. *Journal of Sound and Vibration*. Vol.270, pp.1045-1055, ISSN 0022-460X
- Özdemir, Ö. & Kaya, M.O. (2006). Flapwise bending vibration analysis of a rotating tapered cantilever Bernoulli-Euler beam by differential transform method. *Journal of Sound and Vibration*. Vol.289, pp.413-420, ISSN 0022-460X

- Ozgumus, O. & Kaya, M. O. (2010). Vibration analysis of rotating tapered Timoshenko beam using DTM. *Meccanica*. Vol. 45, pp. 33-42, ISSN 0025-6455
- Rossi R.E. (2007). *Introducción al análisis de Vibraciones con el Método de Elementos Finitos*. EdiUNS, Universidad Nacional del Sur, ISBN 978-987-1171-71-2, Bahía Blanca, Argentina.
- Rossi, R. E.; Gutiérrez R. H. & Laura P. A. A. (1991). Transverse vibrations of a Timoshenko beam of nonuniform cross section elastically restrained at one end and carrying a concentrated mass at the other. *J. Acoust. Soc. Am*, Vol.89, pp.2456-2458.
- Seon, M. H.; Benaroya, H. & Wei, T. (1999). Dynamics of transversely vibrating beams using four engineering theories. *Journal of Sound and Vibration*. Vol.225, pp.35-988, ISSN 0022-460X
- Singh, A.P.; Mani, V. & Ganguli, R. (2007). Genetic programming metamodel for rotating beams, *CMES - Computer modelling in Engineering and Sciences*, Vol.21. No.2, pp. 133-148.
- Shu, C. (2000). *Differential Quadrature and Its Application in Engineering*, Springer-Verlag, ISBN 1852332093, London, England
- Shu, C. & Chen, W. (1999). On optimal selection of interior points for applying discretized boundary conditions in DQ vibration analysis of beams and plates. *Journal of Sound and Vibration*. Vol.222, No.2, pp. 239-257, ISSN 0022-460X
- Vinod, K. G., Gopalakrishnan, S. & Ganguli, R. (2007), Free vibration and wave propagation analysis of uniform and tapered rotating beams using spectrally formulated finite elements. *International Journal of Solids and Structures*; Vol.44, pp. 5875-5893, ISSN 0020-7683
- Yang, J. B.; Jiang, L. J. & Chen, D. CH. (2004). Dynamic modelling and control of a rotating Euler-Bernoulli beam. *Journal of Sound and Vibration*. Vol.274, pp. 863-875, ISSN 0022-460X



Mechanical Engineering

Edited by Dr. Murat Gokcek

ISBN 978-953-51-0505-3

Hard cover, 670 pages

Publisher InTech

Published online 11, April, 2012

Published in print edition April, 2012

The book substantially offers the latest progresses about the important topics of the "Mechanical Engineering" to readers. It includes twenty-eight excellent studies prepared using state-of-art methodologies by professional researchers from different countries. The sections in the book comprise of the following titles: power transmission system, manufacturing processes and system analysis, thermo-fluid systems, simulations and computer applications, and new approaches in mechanical engineering education and organization systems.

How to reference

In order to correctly reference this scholarly work, feel free to copy and paste the following:

Diana V. Bambill, Daniel H. Felix, Raúl E. Rossi and Alejandro R. Ratazzi (2012). Free Vibration Analysis of Centrifugally Stiffened Non Uniform Timoshenko Beams, Mechanical Engineering, Dr. Murat Gokcek (Ed.), ISBN: 978-953-51-0505-3, InTech, Available from: <http://www.intechopen.com/books/mechanical-engineering/free-vibration-analysis-of-centrifugally-stiffened-non-uniform-timoshenko-beams>

INTECH

open science | open minds

InTech Europe

University Campus STeP Ri
Slavka Krautzeka 83/A
51000 Rijeka, Croatia
Phone: +385 (51) 770 447
Fax: +385 (51) 686 166
www.intechopen.com

InTech China

Unit 405, Office Block, Hotel Equatorial Shanghai
No.65, Yan An Road (West), Shanghai, 200040, China
中国上海市延安西路65号上海国际贵都大饭店办公楼405单元
Phone: +86-21-62489820
Fax: +86-21-62489821

© 2012 The Author(s). Licensee IntechOpen. This is an open access article distributed under the terms of the [Creative Commons Attribution 3.0 License](#), which permits unrestricted use, distribution, and reproduction in any medium, provided the original work is properly cited.