Thermo-Mechanical Treatment of Glass-Balloon-Dispersed Metal Matrix Composite

Osamu Umezawa
Yokohama National University
Japan

1. Introduction

Mechanical aspects of recycled composites are demonstrated in this chapter. Aluminium and its alloys are probably the most widely used matrix materials for metal matrix composites (MMCs) (Rodriguez-Castro, 2002). Although reinforcements in the form of continuous and discontinuous fibers have already been investigated in detail (Yu & Lee, 2000), discontinuous reinforcements such as those of dispersoids have also become more popular. Subsequent working of such dispersoid-reinforced MMCs can enhance their mechanical properties (Seah & Hemanth, 2007). Aluminium alloy/glass composites cast with glass particles of sizes in the range of several tens of micrometers have been fabricated and have exhibited superior mechanical properties with regard to strength and wear resistance (Nicom & Nomura, 2006; Gibson & Ashby, 1999).

Since magnesium is the lightest metal in the structural materials, magnesium alloys also have great potential for applications in automotive and aerospace industries. Furthermore, solid-state recycling processes of magnesium alloy chips and/or fragments such as hot extrusion, mechanical alloying, equal channel angular extrusion (ECAE) and cyclically repeated plastic working (RPW) have been developed (Mabuchi et al., 1995; Clark et al., 1997; Kondoh et al., 2003a, 2003b; Kondoh & Luagnvaranaunt, 2003c). The recycled materials show superior mechanical properties such as high strength and high strain-rate super-plasticity. Kondoh et al. (Kondoh & Luagnvaranaunt, 2003c) has established a process to fabricate high performance magnesium composite, in which Mg$_2$Si and MgO particles were synthesized due to deoxidization of SiO$_2$ glass by magnesium, and a refinement of both those dispersoids and magnesium matrix grains was achieved during RPW. Under only 2 mass% SiO$_2$ addition in AZ31 alloy, the Mg$_2$Si compound was employed as a reinforcement of the composite, because of its high hardness, high Young’s modulus and superior corrosion resistance. In order to fabricate the composites of glass balloons and magnesium alloy chips, however, their mixture should be consolidated with the condition for synthesis of Mg$_2$Si and MgO phases.

On the other hand, various porous metals have been fabricated and their specific properties such as ultra-lightweight and low strength have been considered with more attention (Gibson & Ashby, 1999). Further, porous ceramics (e.g., Al$_2$O$_3$, ZrO$_2$, TiO$_2$, SiC, TiC, ZrC, etc.) have been used to develop ceramic matrix composites by various processing routes such as the melt stirring process and pressure infiltration technique (Chou et al., 2000;
Wilkes et al., 2006; Cao et al., 2004a, 2004b; Rambo et al., 2004, 2005). Therefore, the dispersion of porous particles in a metal matrix is one of the methods employed to control the size and distribution of the cell structure in the composite. Recently, new recyclable materials composed of recycled aluminium and glass balloons have been developed (Kashiwaya et al., 1999) and fabricated for building panels (Naigai Building Materials Corporation). However, in order to take the maximum advantage of their specific properties such as ultra lightweight and good electromagnetic shielding for mechanical or electronic parts, a workability that can deform them into any form is required. Since the balloon is a brittle second-phase particle, it is necessary to include either a thermomechanical treatment to plasticize the composite or perform near-net-shape casting. Umezawa et al. (Umezawa & Nagai, 1999) developed a repeated thermomechanical treatment (RTMT) to refine the microstructure of Al-Si cast materials; the materials successfully obtained good plasticity by this treatment. The RTMT was adopted to control the microstructure of the glass-balloon-dispersed aluminium alloy (AC3A) matrix composite in order to achieve a satisfactory stress-strain relationship and ductility for the composite (Shiga & Umezawa, 2007). Furthermore, the mixture of glass balloons and AZ31 magnesium alloy chips was thermomechanically treated to form their composite with Mg$_2$Si and MgO phases.

2. Tensile and compressive properties of glass-balloon-dispersed aluminium alloy composite

A repeated thermomechanical treatment (RTMT) was adopted to control the microstructure of a glass-balloon-dispersed aluminium alloy (AC3A) composite. The RTMT, which involves the repetition of a multi-step process and followed by heat treatment, was applied to a cast plate of the composite. The composite was successfully worked into a rod or sheet by either swaging or flat rolling. The porous glass balloons were deformed or cracked so that the composite could be worked easily. The swaged material exhibited higher tensile strength, Young’s modulus, and elongation as compared to the cast material. The composite materials also exhibited excellent energy absorption properties.

2.1 Microstructural modification by repeated thermomechanical treatment

2.1.1 Test material

The test material (AC) was a 10-mm-thick cast plate (Alcelite™ (Naigai Building Materials Corporation)), which consisted of a recycled glass balloon and aluminium alloy AC3A. The major compositions of the glass balloon were 68 SiO$_2$, 6.3 Al$_2$O$_3$, 0.6 Fe$_2$O$_3$, 0.6 MgO, 9.5 CaO, 11.7 Na$_2$O, and 1.3 K$_2$O in mass%. The major properties of the AC and AC3A are listed in Table 1. Figure 1 shows the microstructure of the AC material and glass balloon. The porous

<table>
<thead>
<tr>
<th>Material</th>
<th>Tensile strength, $\sigma_B$ / MPa</th>
<th>Compressive strength, $\sigma_C$ / MPa</th>
<th>Bending strength, $\sigma_B$ / MPa</th>
<th>Bending elastic modulus, $G$ / MPa</th>
<th>Thermal conductivity, $\kappa$ / W·m$^{-1}$K$^{-1}$</th>
<th>Thermal expansion coefficient, $\alpha$ / $\times 10^{-6}$K$^{-1}$</th>
</tr>
</thead>
<tbody>
<tr>
<td>AC</td>
<td>15.7</td>
<td>49</td>
<td>41.2</td>
<td>144</td>
<td>54.4</td>
<td>16.5</td>
</tr>
<tr>
<td>AC3A</td>
<td>170</td>
<td>170</td>
<td>196</td>
<td>686</td>
<td>217.7</td>
<td>23.6</td>
</tr>
</tbody>
</table>

Table 1. Mechanical and thermal properties of the test material (AC) and AC3A.

www.intechopen.com
glass balloon with a diameter of approximately 1 mm is dispersed in the AC3A matrix. The balloon contains numerous closed bubbles. The specific gravities of the glass balloon and AC3A are $1.58 \times 10^{-4}$ and 2.7 g/cm$^3$, respectively. The specific gravity of the AC material, which was measured with a rectangular bar of dimensions $9.57 \times 9.47 \times 32$ mm, was obtained as 1.5 g/cm$^3$. The volume fraction of the glass balloon was estimated to be approximately half in the AC. The material annealed at 793 K for 1.8 ks was designated as AN.

![Microstructure of the AC material dispersed glass balloons in the AC3A matrix](image1.jpg)

![Whole view of a glass balloon](image2.jpg)

**Fig. 1.** Microstructure of the (a) AC material dispersed glass balloons in the AC3A matrix and (b) whole view of a glass balloon.

### 2.1.2 Repeated thermomechanical treatment and modified microstructure

In order to avoid fracture, the AN samples were worked in multiple steps with intermediate annealing at 793 K for 1.8 ks by using the RTMT process (Umezawa & Nagai, 1999). The samples were machined either as round bars (10 mm in diameter × 150 mm in length) for swaging or as plates (10 mm in thickness × 20 mm in width × 200 mm in length) for flat-rolling. After the samples were worked by either swaging or flat-rolling in multiple steps at room temperature, they were formed into a rod (CS) or sheet (CR). The reduction in section area per working step was approximately 10%, and the working-annealing cycle was repeated for a total section area reduction of approximately 90%. For the resulting RTMT materials in the final state, a multiple-step process involving either swaging or flat-rolling was performed at a different value of total section area reduction. The working strains were defined as $\eta = \ln \frac{A_0}{A}$, where $\eta$ denotes the working strain; $A_0$, the initial section area of the sample; and $A$, the section area of the worked sample. The specific gravities of the worked CR and the selected CS samples ($\eta = 0.46$ and 1.38) were measured and their microstructures were observed. In the final step, the samples with or without annealing were indicated as -H or -A, respectively.

The specific gravity of the materials increases from 1.5 to 2.3 g/cm$^3$ with the working strain up to $\eta = 2.6$ (Fig. 2). This indicates that the relative density of the materials increases. In the case of rolling, a linear relationship is obtained between the working strain and specific gravity. However, the CS material exhibits a higher value of specific gravity for a lower working strain. Although the data plots for the CS material are limited, the glass balloons in
the CS after swaging may be divided and aligned along the working direction with a lower working strain due to a type of hydrostatic stress effect.

![Graph showing variation in specific gravity of test material during swaging or rolling.](image)

**Fig. 2.** Variation in the specific gravity of the test material during swaging or rolling. (Shiga & Umezawa, 2007)

Then microstructures of the composites after swaging or rolling was observed and compared, as shown in Figs. 3 and 4. Rod-like voids in the CS (Figs. 3c and 3d) or pancake-like voids in the CR (Figs. 4c and 4d) are observed at higher working strains. The porous balloons may be cracked and dropped from the sample surface. Comparing the microstructure in the transverse section of the CS and CR materials that were worked at lower working strains, it is observed that the radius of the voids in the CS (Fig. 3b) is lower than that in the CR (Fig. 4b). This observation agrees with the higher value of specific gravity at lower working strains.

### 2.2 Tensile and compressive properties

#### 2.2.1 Tensile properties

A tensile test was performed for the AC, AN, and CS (η = 1.38) materials. Smooth-type specimens with a gauge diameter of 3.5 mm and length of 30 mm were used for the AC and AN materials. For the CS, the specimens were obtained from rods parallel to the longitudinal direction, and the gauge geometry was mechanically polished to obtain a
Thermo-Mechanical Treatment of Glass-Balloon-Dispersed Metal Matrix Composite  

The elongation was monitored by a clip gauge with the knife-edges set onto the tensile specimen. Its gauge length was 25 mm. A crosshead speed of 0.5 mm/min was selected in a motor-driven testing machine at 293 K (in air) under displacement control.

Fig. 3. Microstructure of the CS materials in the longitudinal section (a) and (c) and transverse section (b) and (d): (a),(b) $\eta = 0.46$ and (c),(d) $\eta = 1.38$. (Shiga & Umezawa, 2007)

Fig. 4. Microstructure of the CR materials with working strain (a),(b) $\eta = 0.41$ and (c),(d) $\eta = 1.38$. (Shiga & Umezawa, 2007)
Figure 5 shows the tensile stress-strain curves and Table 2 summarizes the tensile properties. The swaged materials (CS-H and CS-A) exhibit higher values of Young’s modulus as compared to the cast materials (AC and AN). This may be related to the increase in specific gravity due to deformation of the balloons. The tensile strength of the CS materials is approximately thrice that of the cast materials. The RTMT not only increases the specific gravity but also the uniform elongation. The CS materials exhibit necking instability; however, the cast materials are fractured before approaching it. Hence, a microstructural modification by the RTMT could result in the avoidance of early fracture due to tension, which is characteristic of the cast materials. The load drops that appear in the curves of the swaged materials and the early fracture in the cast materials may result from the instantaneous separation between the balloons and the matrix. In fact dimple in the matrix traced the interface with dropped glass-balloon or brittle fractured glass-balloon mostly covers the fracture surface of the AC, as shown in Fig. 6a. The manner in which these fracture occurred was not detected in the CS. The annealing at 793 K for 1.8 ks results in higher uniform elongation in both the AC and CS materials. In particular, the annealed CS material (CS-A) showed an excellent elongation up to a strain of approximately 20%. This manner is typical of materials that undergo the RTMT (Umezawa & Nagai, 1999). Although delamination is detected in the tensile fracture in the CS materials, as shown in Fig. 6b, the RTMT may allow greater flexibility in the working of the composite.

![Stress-strain curves in the tensile test. (Shiga & Umezawa, 2007)](image)

<table>
<thead>
<tr>
<th>Material</th>
<th>0.2% proof stress, $\sigma_{0.2}$ /MPa</th>
<th>Tensile strength, $\sigma_B$ /MPa</th>
<th>Elongation, $\varepsilon_l$ /%</th>
<th>Young’s modulus, $\mu$ /GPa</th>
</tr>
</thead>
<tbody>
<tr>
<td>AC</td>
<td>15</td>
<td>15</td>
<td>1.4</td>
<td>2.2</td>
</tr>
<tr>
<td>AN</td>
<td>14</td>
<td>17</td>
<td>2.7</td>
<td>1.9</td>
</tr>
<tr>
<td>CS-H</td>
<td>35</td>
<td>49</td>
<td>2.7</td>
<td>5.2</td>
</tr>
<tr>
<td>CS-A</td>
<td>27</td>
<td>54</td>
<td>19</td>
<td>5.3</td>
</tr>
</tbody>
</table>

Table 2. Tensile properties of the CS materials. (Shiga & Umezawa, 2007)
The area fraction of the matrix in the transverse section of the CS is higher than that of the AC, and the matrix in the CS continuously exists along the tensile direction in the form of a fiber. Therefore, the CS may exhibit the tensile performance of the aluminium matrix rather than the AC; such behaviour results from the reduction in the stress concentration, which induces fracture of the matrix.

Fig. 6. Fracture surfaces of the (a) AC and (b) CS-A. (Shiga & Umezawa, 2007)

2.2.2 Finite element analysis
Interactive algorithms (Yamada, 2006) used for computing the averaged response of non-linear composites have been applied to microscopically analyse the tensile deformation in the elastic regime. For the AC, spherical glass balloons were installed into the aluminium matrix as a closed-cell foam. Pores were distributed in the balloon. For the CS, ellipsoidal (rod-like) pores were installed into the matrix since the balloon was collapsed into pieces and could be plastically deformed within the matrix.

A simple stress analysis was applied to the materials in order to discuss the fracture manner and stress-strain relationship mentioned in above. Figure 7 shows an analysis map for the principal stress distribution under a microstrain in the elastic regime where the applied stress is 1 MPa in tension along the y axis. The stress concentration appears around the glass balloons. When a glass balloon represents as a sphere, its diameter along the y-axis is taken for the earth’s axis (NS). Then the equator is perpendicular to NS. The glass balloon at the lower left in Fig. 7 represents a reference sphere. The compressive stress with the $\sigma_{xx}$ component is relatively high at both N and S poles (indicated by arrows in Fig. 7a). The principal stress, $\sigma_{yy}$, acts near the interface, especially around the equator. These behaviours agree with the stretching of the faces of the closed-cell foam under tension [6]. Thus, the tensile stress around the equator on the sphere occurs due to tension and may lead to fracture. The stress concentration of the $\sigma_{yy}$ is estimated about three times of the applied stress. On the contrary, the compressive stress around the equator on the neighbouring balloon occurs in Fig. 7b. The maximum magnitude of the $\sigma_{yy}$ in compression is almost the same in tension. These stress concentration phenomena agrees with the fracture surface shown in Fig. 6a. The brittle fractured glass-balloon may result from the concentrated tensile
stress, and the glass-balloon concentrated compressive stress may be separated from the matrix. Even though the interface strength between the glass-balloon and the matrix is very low, it reveals that the stress concentration around the equator on the glass-balloon may predominantly affect the tensile fracture behaviour.

Fig. 7. Principal stresses in the AC material: (a) $\sigma_{xx}$ and (b) $\sigma_{yy}$. Applied tensile stress is 1 MPa and its direction is parallel to the y-axis. The area in white colour shows the pores in the glass balloon. Arrows in (a) indicate a compressive stress concentration part. Arrows in (b) indicate a tensile stress concentration part. (Shiga & Umezawa, 2007)
In the case of the CS material, the stress concentration mostly appears near at both the top and bottom parts of the elongated pore, as shown in Fig. 8. The influence of tensile stress concentration at the ellipsoidal closed-cell foam due to fracture may be lower than that at the spherical foam, although the maximum magnitude of tensile stress concentration in the CS is higher than that in the AC. The tensile stress of $\sigma_{xx}$ between elongated pores occurs. It may give an origin of delamination of the composite.

Fig. 8. Principal stresses in the CS-A material: (a) $\sigma_{xx}$ and (b) $\sigma_{yy}$. Applied tensile stress is 1 MPa and its direction is parallel to the $y$-axis. The elongated pores are indicated in white colour. Arrows in both (a) and (b) indicate a tensile stress concentration part. (Shiga & Umezawa, 2007)
2.2.3 Compressive properties and energy absorption

A compressive test was also performed for the AC, AN, and CS ($\eta = 0.46$) materials. The cylindrical specimens were machined from rods parallel to the longitudinal direction. For the AC and AN materials, the diameter and height of the specimens were 9 mm. For the CS material, the diameter and height of the specimens were 6 mm. In the compressive test, the diameter at the centre of the specimen was measured by using a micrometer. In both the tests, a crosshead speed of 0.5 mm/min was selected in a motor-driven testing machine at 293 K (in air) under displacement control.

Figure 9 schematically illustrates the compressive stress–strain curve of the porous material. The curve generally shows a very low increase in stress (plateau regime); this is followed by an acceleration in the stress (densification). The energy absorbed per unit volume up to a strain at the transition point from the plateau to the accelerated level, $\varepsilon_1$, is defined as (Gibson & Ashby, 1999)

$$ W = \int_0^{\varepsilon_1} \sigma d\varepsilon $$

The absorbed energy is schematically shown by the hatched area in Fig. 9. The energy absorption efficiency, $E$, is also evaluated as

$$ E = \frac{\int_0^{\varepsilon_1} \sigma d\varepsilon}{\sigma_0 \cdot \varepsilon_1} $$

Where, $\sigma_0$ is the stress at strain $\varepsilon_1$.

Fig. 9. Schematic illustration of compressive stress–strain curve in a porous material.

The plateau regime after linear elasticity appears in all the compressive stress-strain curves, as shown in Fig. 10. The arrows in Fig. 10 show the strain of the transition point from the plateau to the accelerated stress, $\varepsilon_1$ for the present evaluation. The higher the strain range of the plateau, the larger is the magnitude of energy absorption. When the material is compressed, the glass balloon within the AC3A matrix is compressed. The working during...
the plateau range may result in the plastic deformation of the matrix and fracture of the glass balloon. However, the balloon is brittle and collapses easily. It may then be harmonized with plastic deformation of the matrix. In fact, the energy absorption efficiency, $E$, of the AC and AN is approximately 0.8–0.9, as listed in Table 3; these values are excellent as they are similar to those of an open-cell foam material (nearly $E = 1$). There is a possibility to use this composite material instead of cell foam materials.

The CS materials show lower inclination of linear elasticity and higher yield stress as compared to the AC and AN materials. Since the CS materials exhibited higher Young’s modulus and relative density, the linear elasticity may involve the effect of bending of the closed-cell. The glass-balloons were broken into pieces and could be easily deformed so that the matrix containing elongated pores might result in the lower modulus. The absorbed energy, $W$, of the CS material is considerably higher than that of the cast material, although the energy absorption efficiency is lower. This may be related to the lower inclination of linear elasticity of the CS.

![Stress-strain curves in the compressive test](https://www.intechopen.com)

Fig. 10. Stress-strain curves in the compressive test. The arrows indicate the strain, $\varepsilon_1$, for each curve. (Shiga & Umezawa, 2007)

<table>
<thead>
<tr>
<th>Material</th>
<th>$\varepsilon_1$</th>
<th>$\sigma_0$ (MPa)</th>
<th>$W$ (MPa)</th>
<th>$E$</th>
</tr>
</thead>
<tbody>
<tr>
<td>AC</td>
<td>0.22</td>
<td>52</td>
<td>9.5</td>
<td>0.8</td>
</tr>
<tr>
<td>AN</td>
<td>0.20</td>
<td>46</td>
<td>8.0</td>
<td>0.9</td>
</tr>
<tr>
<td>CS-H</td>
<td>0.41</td>
<td>109</td>
<td>31</td>
<td>0.7</td>
</tr>
<tr>
<td>CS-A</td>
<td>0.38</td>
<td>143</td>
<td>39</td>
<td>0.7</td>
</tr>
</tbody>
</table>

Table 3. Energy absorption properties. Parameters $W$ and $E$ are the energy absorbed per unit volume as shown in Fig. 9 and the energy absorption efficiency, respectively.

### 3. Fabrication of glass-balloon-dispersed magnesium alloy composite

The mixture of glass-balloon and AZ31 magnesium alloy chips with 1:1 in volume was heated and extruded to form a brick of composites. The bricks were heat-treated in $\alpha$ phase region, and their microstructure was characterized. As the heating time increased, oxidation
of magnesium was accelerated. As heat-treatment temperature was higher the oxidation was also promoted. The heat-treatment condition at 873 K - 1h was chosen to form Mg$_2$Si. As press load increased, the volume fraction of Mg$_2$Si was increased.

### 3.1 Extrusion of magnesium alloy chip and glass balloon mixture

#### 3.1.1 Test materials

Machine chips of AZ31 magnesium alloy and 50 mol% porous glass-balloons were mixed by rocking-mill with 60 rpm for 600 sec. The major compositions of the glass balloon were 68 SiO$_2$, 6.3 Al$_2$O$_3$, 0.6 Fe$_2$O$_3$, 0.6 MgO, 9.5 CaO, 11.7 Na$_2$O, and 1.3 K$_2$O in mass%, and its diameter was approximately 1 mm. The balloon contains numerous closed bubbles. The specific gravities of the glass balloon and AZ31 are $1.58 \times 10^{-4}$ and 1.78 g/cm$^3$, respectively. The mixture was heated in the container, and extruded into cylindrical form with 20 mm in diameter by 100 ton hydraulic press machine under the conditions listed in Table 4.

<table>
<thead>
<tr>
<th>Sample</th>
<th>Weight /g</th>
<th>Heating temperature /K</th>
<th>Start temperature /K</th>
<th>Finish temperature /K</th>
<th>Pass load /ton</th>
</tr>
</thead>
<tbody>
<tr>
<td>No. 1</td>
<td>15</td>
<td>300</td>
<td>300</td>
<td>300</td>
<td>4</td>
</tr>
<tr>
<td>No. 2</td>
<td>15</td>
<td>418</td>
<td>403</td>
<td>393</td>
<td>6</td>
</tr>
<tr>
<td>No. 3</td>
<td>15</td>
<td>468</td>
<td>456</td>
<td>451</td>
<td>14</td>
</tr>
<tr>
<td>No. 4</td>
<td>18</td>
<td>468</td>
<td>456</td>
<td>448</td>
<td>14</td>
</tr>
<tr>
<td>No. 5</td>
<td>18</td>
<td>488</td>
<td>481</td>
<td>472</td>
<td>15</td>
</tr>
<tr>
<td>No. 6</td>
<td>18</td>
<td>538</td>
<td>526</td>
<td>522</td>
<td>15</td>
</tr>
<tr>
<td>No. 7</td>
<td>18</td>
<td>568</td>
<td>553</td>
<td>544</td>
<td>20</td>
</tr>
<tr>
<td>No. 8</td>
<td>18</td>
<td>588</td>
<td>577</td>
<td>567</td>
<td>20</td>
</tr>
<tr>
<td>No. 9</td>
<td>18</td>
<td>638</td>
<td>618</td>
<td>608</td>
<td>20</td>
</tr>
<tr>
<td>No. 10</td>
<td>18</td>
<td>588</td>
<td>577</td>
<td>567</td>
<td>10</td>
</tr>
<tr>
<td>No. 11</td>
<td>18</td>
<td>588</td>
<td>577</td>
<td>567</td>
<td>15</td>
</tr>
</tbody>
</table>

Table 4. The extruded samples from the mixture of AZ31 alloy chip and glass balloon

#### 3.1.2 Macrostructure and microstructure of preform

In appearance a brick of composite was formed into a cylinder at the temperature over 400 K as shown in Fig. 11, but the heating over 573 K was needed to be almost no chipping at the edge of cylinder. Figure 12 shows the macrostructure of the composite in the longitudinal cross section of No.8 sample. The glass-balloon was broken into pieces and dispersed along plastic flow. Density of glass pieces was relatively higher in the vicinity of sample surface, as the sample was extruded at lower temperature.

Figure 13 shows microstructure of the section for No. 9 sample. Grain size was finer in the vicinity of the sample surface and boundary, and pieces of glass balloon were dispersed in
the matrix. Elemental analysis by EPMA demonstrated that fragmented glass-balloons were also distributed not only at the particle boundaries but also in the inside of garins as shown in Fig. 14, and that enriched oxygen was detected at the boundaries. Therefore the boundaries resulted from the interface of AZ31 magnesium alloy chips, and fragmented glass-balloons were partially contaminated there.

![Sample side view (22 mm in height) (a) and bottom view (b) of No. 8](image)

Fig. 11. Sample side view (22 mm in height) (a) and bottom view (b) of No. 8

![Macrostructure of No. 8 in the longitudinal section](image)

Fig. 12. Macrostructure of No. 8 in the longitudinal section
3.2 Solid-state synthesis of Mg$_2$Si and MgO

The DSC (differential scanning calorimeter) curves for the compacts in employing the elemental AZ31 and SiO$_2$ glass powder mixture expect the exothermic peaks at 923 K due to the latent heat of magnesium melting and at 844 K (SiO$_2$ glass transition temperature) due to the phase transformation from $\alpha$ to $\beta$ (Reynard et al., 1996). However, they showed the starting temperature of the exothermic at 730 K (Kondoh & Luagnvaranaunt, 2003c). Therefore, the deoxidizing SiO$_2$ by magnesium has a large possibility to form Mg$_2$Si and
MgO by the reaction in eq.(3), although the ignition temperature in DSC curve may depend on the compositions and crystalline in the glass-balloon.

\[
4\text{Mg} + \text{SiO}_2 \rightarrow \text{Mg}_2\text{Si} + 2\text{MgO}
\]  

(3)

### 3.2.1 Heat treatment of preform

The samples were heat-treated in vacuum at the temperature of 773 K, 823 K, 873 K and 898 K for 1 hr. The heat-treated materials were characterized by X-ray diffraction (XRD) with Cu-Kα1 radiation operated at 40 kV - 200 mA (range: 20-90 deg, step: 0.02, scan: 2 deg/min) and micro-Vickers hardness (load: 1 kg, holding time: 15 s).

In the sample No. 2, a solid-state synthesis was available at the temperature heating over 873 K, and not at lower temperatures due to oxidation of magnesium chips. The isothermal treatment at higher temperature may cause a partial remelting of magnesium α phase. Then the No. 2 samples were heated at the temperatures of 873 K and 898 K for longer time to identify the reacted phases. Figure 15 summarized the estimated volume fractions of crystalline phases formed by isothermal heating, which were calculated from the integrated intensity of peaks in XRD profiles. As the heating time increased, the oxidation of magnesium was accelerated as well as deoxidization of SiO₂. As heat-treatment temperature was higher, the oxidation was also promoted. As a result, the heat-treatment condition at 873 K for 1 h was chosen to form Mg₂Si with less MgO.

![Fig. 15. Effect of heat treatment on estimated volume fractions of crystalline phases for No.2 sample](http://www.intechopen.com)

### 3.2.2 Effect of press load on preform

The samples extruded at about 573 K, i.e. No. 8, 10 and 11, were pressed with different load from 10 to 20 ton, so that the higher press load resulted in finer dispersion of glass balloons.
and higher adhesion of matrix. Figure 16 showed the estimated volume fractions of crystalline phases after heating at 873 K for 1 h. The less deoxidization of SiO$_2$ may be due to smaller faction of interface between SiO$_2$ and magnesium, as the lower press load is. The No. 8 sample of pressed with load 20 ton exhibited much higher volume fractions of Mg$_2$Si and MgO and showed 400 HV after the heat treatment.

![Figure 16](image1.png)

**Fig. 16.** Effect of press load on the volume fraction of crystalline phases after heating at 873 K for the extruded samples at about 573 K

![Figure 17](image2.png)

**Fig. 17.** Effect of press load on the hardness after heating at 873 K for the extruded samples at about 573 K

4. Conclusion

Thermomechanical treatments were adopted to control the microstructure of porous glass-balloon-dispersed aluminium alloy and magnesium alloy composites. The mechanical properties and the microstructure of the composite were characterized.
The porous glass-balloon-dispersed AC3A aluminium alloy composite, which was subjected to the repeated thermomechanical treatment along with the working by either swaging or flat rolling, was successfully formed into a rod or sheet. Microstructural modification by the treatment avoided early fracture due to tension and resulted in good ductility. The swaged material exhibited higher tensile strength and Young's modulus as compared to the cast material. A stress concentration around the glass balloons was confirmed. The stress may lead to fracture. The composite materials also exhibited excellent energy absorption properties. The swaged material showed lower inclination of linear elasticity and higher yielding stress as compared to the cast material.

A new process for fabricating magnesium composites with a solid-state reaction by glass balloons has been proposed. The mixture of glass balloons and AZ31 magnesium alloy chips was thermomechanically treated to form their composite and applied to solid-state synthesis of Mg$_2$Si and MgO phases. It demonstrated a possibility to employ SiO$_2$ glass balloons to fabricate a lightweight magnesium alloy composite.

5. Acknowledgment

The author thanks S. Shiga, Y. Kosuge and Dr. S. Morooka for their help on the experimental works.

6. References


Rambo, C.R., Cao, J., Rusina, O. & Sieber, H. (2005), Manufacturing of biomorphic (Si, Ti, Zr)-carbide ceramics by sol-gel processing, Carbon, Vol. 43, No. 6, (May 2005), pp. 1174-1183, ISSN 0008-6223


Composite materials, often shortened to composites, are engineered or naturally occurring materials made from two or more constituent materials with significantly different physical or chemical properties which remain separate and distinct at the macroscopic or microscopic scale within the finished structure. The aim of this book is to provide comprehensive reference and text on composite materials and structures. This book will cover aspects of design, production, manufacturing, exploitation and maintenance of composite materials. The scope of the book covers scientific, technological and practical concepts concerning research, development and realization of composites.

How to reference
In order to correctly reference this scholarly work, feel free to copy and paste the following:
